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### Identifying the driving processes of Diesel spray injection through mixture fraction and velocity field measurements at ECN Spray A

#### Abstract

Diesel spray mixture formation is investigated at target conditions using multiple diagnostics and laboratories. High-speed Particle Image Velocimetry (PIV) is used to measure the velocity field inside and outside the jet simultaneously with a new frame straddling synchronization scheme. The PIV measurements are carried out in the Engine Combustion Network Spray A target conditions, enabling direct comparisons with mixture fraction measurements previously performed in the same conditions, and forming a unique database at diesel conditions. A 1D spray model, based upon mass and momentum exchange between axial control volumes and near-Gaussian velocity and mixture fraction profiles is evaluated against the data. The 1D spray model quantitatively predicts the main spray characteristics (average mixture fraction and velocity fields) within the measurement uncertainty for a wide range of parametric variations, verifying that a Diesel spray becomes momentum controlled and has a Gaussian profile. A required input to the model is the jet angle, which is obtained experimentally. Although an expected result for a gas jet, this is the first time that combined datasets of velocity and mixture fraction have been obtained in vaporizing sprays at Diesel conditions (900 K, 60 bar). Finally, these results show that a consistent database can be built using advanced diagnostics performed by different institutions when the boundary conditions are well known as prescribed by the ECN Spray A framework.

#### Introduction

Designing fuel-efficient, low-exhaust-emission Diesel engines requires a detailed understanding of the physical phenomena occurring in the combustion chamber. One of the decisive processes in conventional direct-injection Diesel engines is the mixing between air and fuel. Indeed, for fixed operating conditions (cetane number, temperature, density, EGR rate), the turbulent mixing rate is a key factor governing the chemistry and affecting different stages of the combustion process as:

- Ignition delay and premixed flame: an increase of the mixing rate will lead to a decrease of the ignition delay, and thus will affect the premixed flame process and the level of noise
- Controlled diffusion: various studies, in combustion vessels
  [1–3] or in engines[4–9], have shown a correlation between
  the fuel-air ratio at lift-off length and the amount of soot
  produced in the spray. Lowering the fuel-air ratio by
  enhancing the mixing rate is thus desirable in order to
  decrease soot emissions;
- End of injection: Musculus [10] has observed a rapid decrease of mixture fraction near the nozzle tip after the end of injection, later explained by the concept of an "entrainment wave" [11]. This entrainment wave has now been observed experimentally [12], and it may affect

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unburned hydrocarbon emissions (UHC) via over- or under- mixing. It can also have an effect on the ignition delay if the End Of Injection (EOI) occurs before ignition [13].

The fuel-air mixing processes of vaporizing Diesel sprays have been investigated by a combination of experiments and computational efforts [14-21], trying to define the role of large and small scale structures in the air entrainment processes. In [14, 16, 20, 21], air entrainment is supposed to be caused by air engulfment in large structures. However, in [17-19], small scale turbulent structures are described as the main process controlling air entrainment. An approach combining small and large scales has been proposed in [15]. In spite of the uncertainty in the mechanisms governing air entrainment, simplistic models for mass and momentum exchange in one dimension along the axis of the spray have been proposed to compute the spray development and the air-fuel mixing [22, 23]. Although these models show good abilities to predict the behavior of a Diesel spray in terms of vapor penetration, the underlying assumptions concerning mixing processes have not been experimentally validated. Such a validation is of high priority, since it can confirm or expose the hypothesis of the 1D model concerning the physical phenomena driving the mixing process.

However, the study of the air entrainment in Diesel jets is of high complexity because of the strong coupling that exist between the contributing mechanisms, such as fuel evaporation, turbulence, small scale mixing, and so forth. Ideally, the coupled quantitative measurement of several parameters such as flow velocities and mixing characteristics would be required. These types of measurements are routine in ambient temperature gas jet environments [24], but remain challenging in a Diesel environment, where the high-pressure, high-temperature gradients and the presence of liquid droplets make the application of optical diagnostics more complicated.

To overcome these difficulties, a group of international laboratories joined efforts to form the Engine Combustion Network (ECN, see website https://ecn.sandia.gov/) to build a collaborative database on engine combustion, and in particular on Diesel sprays. The ECN seeks to bring a better understanding of the physical phenomena occurring during injection and combustion, and to provide a unique and complete database for CFD model validation. One of the main requirements for the construction of such a database is the definition of a common target condition that needs to be reproduced by each participating institution to leverage research capabilities and advanced diagnostics of all ECN participants. The first target condition, called Spray A has been defined with detailed ambient (900 K, 60 bar, 22.8 kg/m3, 0% or 15% oxygen) and injector (common rail, 1500 bar, KS1.5/86 nozzle, 0.090-mm orifice diameter, n-dodecane, 363 K) conditions [25]. Establishing and improving Spray A conditions in unique facilities throughout the world represents a major step forward in the establishment of highquality, quantitative data sets for engine spray combustion. Boundary

conditions and basic Spray A characterization based on standardized diagnostics have been extensively compared [26–28], showing the ability of different laboratories to reproduce the same boundary conditions. Based on this preliminary validation, mixture fraction fields have been obtained by Sandia National Laboratories (SNL) [29], and velocity fields by IFPEN [27], for identical boundary conditions and with nominally identical injectors. Therefore, the velocity and mixture fraction fields, although obtained in 2 different facilities, should be representative of the same mixing process. This assumption needs to be verified, in order to validate the scientific approach of the ECN network

This requires the use of modeling. CFD codes could be used to answer this question, but at the cost of a careful, and therefore long, calibration of the turbulence model, and of a long computational time. The 1D spray model mentioned earlier [11, 22, 23, 29, 30], has proven to accurately match the spray tip penetration with respect to time. It is based on assumptions (see section "Description of the 1D model" for details) that induce specific velocity and mixture fraction fields within the spray. However, those estimated fields have so far not been compared to experimental ones, mainly because the required data for such a comparison, velocity and mixture fraction fields, are difficult to obtain and are therefore rare. Here, the environment of the ECN provides the opportunity to have access to these data for similar boundary conditions and thus to validate the behavior of the 1D spray model against experimental data. This is the goal of the present paper.

As an answer to the above questions, the objectives of this work are:

- a validation of the 1D spray model against experimental data (velocity and mixture fraction fields in the spray). This step will validate the underlying assumptions concerning the physical phenomena leading the mixing processes;
- a validation of the ECN scientific approach, by the comparison of the velocity and mixture fraction fields within the spray. If proven successful, results obtained by different institutions may be combined to form a unique and comprehensive database for model validation. This comparison will be performed using the 1D spray model.

The experimental apparatus and experimental techniques used to obtain the velocity fields are first presented (mixture fraction fields have been obtained by Sandia, thus the experimental setup is available in [29] and will not be presented here). Then, the experimental results are analyzed to provide information on the driving processes of mixture formation, and to validate the assumptions of the 1D spray model. At last, a comparison of the velocity and mixture fraction fields is performed in order to assess whether they can be combined in a unique database that can be used for CFD model validation.

#### Experimental apparatus and methods

#### IFPEN Constant-Volume Preburn chamber

Within this work, the IFPEN Constant-Volume Preburn (Figure 2) chamber is used to study fuel sprays at engine-relevant conditions. The ambient pressure, temperature and species at the time of injection are varied by igniting a premixed combustible-gas mixture that burns to completion. Following the spark-ignited, premixed combustion, the combustion products cool over a relatively long time ( $\sim 1$  s), due to heat transfer to the vessel walls and as a result the

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vessel pressure slowly decreases. When the desired pressure and temperature are reached, the diesel fuel injector is triggered and fuel injection occurs. More detailed information can be found in [25, 27].



Inner width : 125mm Volume : 1400cm<sup>3</sup> Windows diameter : 80mm Wall temperature: 473K Nozzle tip protrusion: 3mm 4 sparks plugs in the lower corners

Figure 1 - Global view (left), cross-section (center, with an injector mounted on top), and characteristics (right) of IFPEN constant volume vessel

#### **Boundary conditions**

Non-reacting spray A boundary conditions have been targeted for this work, along with two mono-parametric variations of the injection pressure (1000bar instead of 1500bar), and of the ambient density (15.2kg/m3 instead of 22.8kg/m3). These variations have been selected because they directly affect the mixing processes. ECN injector number 210678-aged has been used. All the boundary conditions are summarized in Table 1, both for the IFPEN setup and for the Sandia one. The values in brackets represent the monoparametric variations performed around the standard spray A conditions. Even if the ECN injectors are nominally identical in order to allow different laboratories to perform experimental campaigns simultaneously, some differences in the orifice diameter and in the behavior such as the mass flow rates and momentum have been observed. As detailed in [31], it leads to differences in spray development, ignition and combustion process. These differences can be explained either by slightly different machining at production, or by an alteration of the orifice surface of the injector caused by its utilization, as highlighted in [32]. For the present study, it is worth mentioning that injectors 210677 and 210678-aged show different values for mass flow rate and momentum, which are likely to impact the mixing processes. These flow characteristics are summarized in Table 1. It must be noted that the momentum measurement for injector 210678-aged was performed while it had already been used for a while. Aging (oxidation of nozzle to reduce diameter and lower flow rate) was discovered in experiments immediately after the PIV measurement, indicating that aging had occurred (https:// ecn.sandia.gov/diesel-spray-combustion/target-condition/injectorwhereabouts/injector-history/?nam=210679.#210679). Consequently, earlier measurements of discharge coefficient, area contraction coefficient, and nozzle diameter for injector 201678 [31] do not apply. We therefore refer to this nozzle as "201678-aged."

Table 1. Boundary conditions of the experiments performed at IFPEN (velocity fields obtained by PIV) and Sandia (mixture fraction fields obtained by Rayleigh scattering)

		IFPEN	SANDIA
Density	kg/m <sup>3</sup>	22.8 [15.2]	22.8 [15.2 7.6]
Temperature	K	900	900
Injection Pressure	bar	1500 [1000]	1500 [1000]
Injection Duration	μs	1500	6000
%O <sub>2</sub>	%vol	0	0
Fuel	-	n-dodecane	n-dodecane
Injector reference	-	210678-aged	210677
Mass flow rate	g/s	2.25	2.27
Diameter	μm	88.6	83.7
Momentum	N	1.22	1.46
Spray Angle	0	20.8	21

#### PIV experimental setup

The velocity fields are obtained through time-resolved Particle Imaging Velocimetry at 10kHz, whose experimental setup is presented in Figure 2. The beam of a double-pulsed 532nm Nd:YAG (Quantronix Condor, lamp pumped) laser is formed into a laser sheet by the combination of a spherical (f=300mm) and a cylindrical (f=-100mm) lens. The focal lengths of the lenses are selected in order to obtain a laser sheet the same width as the entrance window (80mm). The laser sheet intercepts the spray axis, in the focal plane of the camera as the injector is horizontally mounted. The two laser pulses are separated by 4µs, and the energy of each pulse is approximately 2.3mJ. The particles used for PIV measurements are porous SiO2 powder (Seika Corporation, B-6C), with a density  $\rho_p=600$ kg/m3 and an average diameter  $d_p=2\mu m$ . They can resist the preburn event necessary to reach high temperatures, and are seeded into the ambient gases during the filling process of the vessel [33].. The ability of these particles to be used as tracer of the flow will be discussed in the "Results" section. A Photron SA1 high-speed camera equipped with a Nikkor 50mm f/1.4 lens is used to collect the Mie scattering on the particles in the direction orthogonal to the laser sheet. The size of images is 704x240 pixels, corresponding to 75.2x25.6 mm (with a resolution of 9.4 pixel/mm). The images are processed with the LaVision Davis software, using a multi-pass processing with a first pass with a 64x64 pixels interrogation window and a second pass with 8x8, 16x16 or 32x32 pixels interrogation window, and 50% overlap. Vectors corresponding to peak ratio lower than 1.1 are deleted. To achieve statistically meaningful results, the velocity fields are averaged over 10 individual injection events. The convergence of the velocity fields will be evaluated in the Results section The uncertainty of the results, referring to the 95% confidence interval computed using the Student's t-distribution, will be shown on the results. The peak locking phenomena occurs when particles are smaller than one pixel and is characterized by an over-population of particle displacements taking integer values [34]. It has been estimated in a ROI corresponding to the spray region (see red rectangle on the instantaneous velocity field in Figure 2). The peak lock value computed by the software is lower than 0.1, which indicates that the peak locking effect is acceptable.

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Figure 2 - a) Experimental setup for PIV measurements. b) Example of raw PIV image and related instantaneous velocity field and an ensemble average velocity field. The red rectangle on the instantaneous velocity field indicates the ROI used to estimate peak locking.

#### Description of the 1D model

The 1D Diesel spray model has been extensively detailed in the literature [11, 23, 29]. It is based on the main following assumptions:

• the spray is self-similar, with radial shape functions

$$u_x(x,r) = u_{x,c}(x) \left[ 1 - \left(\frac{r}{R(x)}\right)^{\alpha} \right]^2$$
  
$$\xi(x,r) = \xi_c(x) \left[ 1 - \left(\frac{r}{R(x)}\right)^{\alpha} \right]^2 \tag{1}$$

as quasi-Gaussian radial profiles for axial velocity  $u_x$  and fuel volume fraction  $\xi$ , where R(x) is the radius of the spray at axial distance x,  $u_{x,c}(x)$  the axial velocity on the spray axis at distance x, and  $\xi_c(x)$  the fuel volume fraction on the spray axis at distance x. The parameter  $\alpha$  defines the shape of the radial profiles. Usually, the value of  $\alpha$  has been set to 1.5 [11, 35].

 the vapor angle is constant. The angle is a crucial input parameter of this model that accounts for detailed mixing and entrainment phenomena. Indeed, the exact phenomena occurring at micro-scale level are still not fully understood, but they result in a certain spray cone angle. This is the only input that is not directly linked to the boundary conditions and that requires tuning.

- fuel mass flow rate and total momentum are conserved through any cross section of the spray. This means that there is an exchange of momentum between the fuel and the air, and thus that this momentum exchange is the base mechanism of air entrainment.
- With all those assumptions, it is possible to obtain a spatial distribution of axial velocity and fuel volume fraction in the spray. But as mentioned, this requires that the spray angle is known. Following ECN recommendations, the spray angle is derived through vapor penetration measurements [29].

#### Results

#### Validation of the PIV technique

Several factors can affect the reliability of the velocity measurements performed with the PIV technique. As the injected fuel is liquid, some fuel droplets can be mistaken for particles. But, for the studied conditions, the liquid length is approximately 10mm, so liquid droplets are unlikely to be present downstream 15mm. However, among others, the inertia of the particles, the time delay  $\Delta t$  between two consecutive laser pulses and the windowing size during PIV cross-correlation analysis have been identified as possible sources of errors. Also, the statistical convergence of the velocity fields on the 10 injection events is assessed. These limitations are discussed in the following paragraphs.

#### Particles as flow tracers

The Stokes number  $S_k$  is usually used as a criterion to assess whether or not particles are relevant tracers of a given flow.  $S_k$  is defined as the ratio of the characteristic time of the particles  $\tau_p$  to that of the flow  $\tau_f$ , or in other words as the ratio of the acceleration of the particle to that of the surrounding fluid. The characteristic time of the particle  $\tau_p$  can be estimated through the equation of the motion of a spherical particle into an unsteady fluid. This equation has first been given by Basset [36] and later discussed by Melling [37] and Mei [38] for PIV applications. In the case of a high density ratio between the seeding particles and the ambient flow, as shown in [37], the equation simplifies to:

$$\frac{du_p}{dt} = -\frac{1}{\tau_p} \left( u_p - u_f \right) \tag{2}$$

where  $u_p$  [m/s] is the particle velocity,  $u_f$  [m/s] is the flow velocity.  $\tau_p$  is given by:

$$\tau_p = \frac{4}{3} \frac{1}{C_d R e_p} \frac{\rho_p d_p^2}{\mu} \text{, with } R e_p = \frac{\rho_f |u_p - u_f| d_p}{\mu}$$
(3)

where  $C_d$  [-] is the drag coefficient,  $Re_p$  [-] the particle Reynolds number,  $\mu$  [Pa.s] the fluid's dynamic viscosity,  $\rho_p$  [kg/m3] the particle's density and  $d_p$  [m] its diameter, and  $\rho_f$  [kg/m3] the fluid's density.

For  $Re_p$  lower than approx. 1, the Stokes' drag law applies and  $C_d$  can be estimated by  $C_d=24/Re_p$ .

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This is a conservative assumption since it underestimates the real value of  $C_d$  when  $Re_p > 1$ . This ultimately leads to the following formula for  $\tau_{p:}$ 

$$\tau_p = \frac{\rho_p d_p^2}{18\mu} \tag{4}$$

The ambient gases are a mixture of N<sub>2</sub>, CO<sub>2</sub> and H<sub>2</sub>O, having a dynamic viscosity  $\mu \approx 4.10^{-5}$ Pa.s at 800K. This leads to  $\tau_p=3\mu s$ .

Concerning the flow, it can be decomposed into its average and turbulent components. Each of them has a different characteristic time  $\tau_f$ . However, in this study, since the only the average velocity fields are considered, the characteristic time of interest is the one corresponding to the average flow. Considering the average flow, its characteristic time can be inferred from the particulate derivative of the flow, considering only the axial component  $u_x$  of the velocity (along the x-axis of the spray) :

$$\overrightarrow{u_f} = u_x \cdot \overrightarrow{x} \text{ gives } \frac{du_x}{dt} = \frac{\partial u_x}{\partial t} + u_x \frac{du_x}{dx}.$$
 (5)

If we consider the steady-state regime of the flow, we have:

$$\frac{\partial u_x}{\partial t} = 0 \text{ and then } \frac{du_x}{dt} = u_x \frac{du_x}{dx} = u_x \cdot \frac{1}{\tau_f}$$
(6)

Using the 1D spray model to derive the velocity field, its derivative along the x-axis can be computed and we can therefore obtain a 2D map of  $\tau_{f_2}$  and then a 2D map of the Stokes number. The logarithm in base 10 of this Stokes number is plotted in Figure 3. It appears that  $S_k<1$  on the whole 1D spray, and is lower than 0.01 for axial distances greater than 20mm. This indicates that the particles are relevant tracers for the average flow in the spray.



Figure 3 - 2D map of log10(S<sub>k</sub>) based on  $\tau_p$  estimated from the particles characteristics and  $\tau_f$ , extracted from the 1D-spray model based flow acceleration

#### Convergence of the velocity fields

As explained in the description of the experimental setup, velocity fields are ensemble averaged on 10 injections. The obtained results are presented in Figure 4, for a delay of  $4\mu$ s, an interrogation window of 16x16 pixels and a timing corresponding to 1.6ms after SOI. The convergence of the resulting ensemble averaged velocity fields must be assessed. This is done by looking at the convergence of the velocities at 4 positions arbitrarily selected on the spray axis, at 20mm, 30mm, 40mm and 50mm from the orifice of the injector, represented by gray cricles. It appears that the convergence of the velocity fields is not satisfying, since the confidence intervals are between 10% and 20% of the corresponding average velocity. This value is considered too high to allow a proper comparison between the measured velocity fields and the ones estimated with the 1D spray model.



Figure 4 – Top three images: evaluation of the statistical convergence of the ensemble averaged velocity fields, for a delay of  $4\mu$ s, an interrogation window of 16x16 pixels and a timing corresponding to 1.6ms after SOI. Bottom two images: evolution of the confidence interval normalized by the average velocity, for the locations corresponding to the gray circles.

To overcome this limitation, and since for this study only the steadystate average velocity fields are of interest, reconstructed steady-state velocity fields can be computed. At each axial location, the radial profiles of the axial velocity  $u_x$  from the ensemble average velocity field and at all the timings are considered. For velocity fields spanning from SOI to EOI+2ms, at 10kHz, this gives 36 radial profiles at each axial location. For x=39.7mm, these profiles are plotted on Figure 5. All the profiles whose maximum value is greater Page 5 of 12

than a certain percentage of the highest value are considered to correspond to steady-state timings. This percentage has been arbitrarily set to 80%. For the case presented in Figure 5, the velocities are in steady-state between 0.9ms and 1.8ms after SOI. During this time range, no systematic trend on the maximum value of the velocity is observed, which confirms that the steady-state has been truly identified. This analysis is performed for all the axial distances, allowing to identify the steady-state timings at each axial location. Then, these steady-state timings are used to select the steady-state profiles on the 10 injections events. Thus, at each axial location, instead of having only 10 profiles to compute the ensemble average, there are 10 multiplied by the number of steady-state timings, which leads to a better statistical convergence. These profiles are plotted in Figure 5, bottom, for the case x=39.7mm. It appears that, although the standard deviation  $\sigma$  is quite high, the uncertainty u is about 5% of the mean value since it is computed on 100 profiles.



Figure 5 - Top: ensemble averaged radial profiles of the axial velocity, for all the timings between SOI and EOI+2ms. Red profiles are considered to be representative of the steady-state, gray ones are considered transient. Bottom: Same as the top figure, for all the instantaneous radial profiles. The mean  $\mu$ , standard-deviation  $\sigma$ , and uncertainty u are plotted respectively in plain, dotted and dashed black lines.

Using this methodology, the convergence of the reconstructed steadystate velocity fields is presented in Figure 6. Figure 6 is similar to Figure 4, but for the reconstructed steady-state velocity field. In this

case, it appears that the confidence intervals are about 5% of the average velocity, which is much lower than the value plotted in Figure 4. In the following, all the velocity profiles that are discussed are extracted from the reconstructed steady-state velocity fields.



Figure 6 – Top three images: Evaluation of the statistical convergence of the instantaneous velocity fields, for a delay of 4µs and an interrogation window of 16x16 pixels after SOI. Bottom two images: evolution of the confidence interval normalized by the average velocity at the locations indicated by the grey cross of the left hand side images.

#### Delay ( $\Delta t$ ) between laser pulses

Another possible source of error is the delay  $\Delta t$  between two consecutive laser pulses. If this delay is too large, the movement of the particles between two consecutive laser pulses is too large, and the PIV correlation algorithm fails to determine the correct displacement. On the opposite, if it is too short, the displacement is

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too small to be accurately determined compared to the image resolution. The former case is an issue when considering the high velocities near the nozzle tip. In order to assess the impact of this delay, a variation of its value has been performed, in the non-reacting Spray A conditions. In addition to the standard delay of  $4\mu$ s, delays of  $10\mu$ s and  $20\mu$ s have been tested. Figure 7 shows the centerline axial velocities corresponding to these different delays, for an interrogation window size of  $16 \times 16$  pixels.



Figure 7 - Influence of the delay  $\Delta t$  between 2 laser pulses on the measured axial velocity for an interrogation window size of 16x16 pixels (top), and on the validity of the computed vectors (bottom three images)

It appears that, in the very far field (x>50mm), the axial velocities are similar for the three values of  $\Delta t$ . This tends to prove that, in that region, the measurements are not dependent upon the value of the delay between the laser pulses. However, for shorter distances, Figure 7 shows that higher  $\Delta t$  are not able to resolve high velocities. A  $\Delta t$  of 20 µs permits measurement of velocities lower than 30m/s. For higher velocities (x<50mm), the measurements are diverging from the 2 other values of  $\Delta t$ . The same can be observed for  $\Delta t=10\mu s$ : below 40mm (i.e. velocities higher than 40m/s), the measurements are not relevant. For the delay used in this study ( $\Delta t=4\mu s$ ), there is also a maximum value for the velocities that can be resolved. And it is probable that the unphysical drop in velocity observed below 20mm is caused by the existence of this limitation, as the velocities tend to be higher near the nozzle tip. This can be confirmed by the statistics on the validity of the computed vectors, shown on the 3 bottom images in Figure 7. As explained in the description of the PIV technique, vectors whose peak ratio is lower than 1.1 are deleted. Therefore, for each location, the numbers of valid vectors in the ensemble average can be counted. This is shown in Figure 7. It appears that when the delay increases, the number of valid vectors decreases. The most favorable case is thus obtained with a delay of 4µs. But even in this case, the number of valid vectors drops for x<20mm on the spray axis, which confirms that the velocities computed in this region are not reliable. From a theoretical point of view, to allow a correct detection, the time interval  $\Delta t$  must respect the following criteria [39]:

$$\Delta t < 0.25. d_I / |u| \tag{7}$$

where  $d_i$  is the size of the interrogation window. In our case, knowing  $\Delta t$  and  $d_i$ , we can compute the theoretical maximum value of the velocity that can be computed. This gives 106m/s, 45m/s and 30m/s respectively for  $\Delta t$  4µs, 10µs and 20µs. This is in agreement with the empirical values we have measured (70m/s, 40m/s and 30m/s). This means that the limitations in velocity measurement observed when varying  $\Delta t$  are governed by the respect of the criteria in Eq. 6. Only for the lower value of  $\Delta t = 4\mu s$  the measured maximum velocity (70m/s) is lower thant the expected one (106m/s). However, in this case, the radial velocity gradients are higher, and can affect the measurement. This will be discussed below. Thus, based on what is seen on Figure 7, velocity measurements will not be considered to be reliable for distances lower than 20mm.

#### Window size for cross-correlation processing

The size of the interrogation window defined for the processing of the PIV results can also have an impact on the measurements. Figure 8 shows the axial velocities obtained with 3 different values of interrogation window. It appears that, downstream of 40mm, all the profiles are similar, therefore we can conclude that the velocities are not significantly affected by the size of the window in this region of the jet. However, below 40mm, some discrepancies appear between the 32x32 results and the other 2. Below 30mm, the same deviation appears between the 8x8 and 16x16 results.

The differences in the resulting velocity fields processed with different windows size could be caused by the velocity gradients existing inside the interrogation windows. This has been studied by Keane and Adrian [40] and also discussed by Westerweel [41]. They have shown that the velocity gradients inside an interrogation window can be neglected when the following condition is satisfied:

$$|a| \ll d_{p}$$
, with  $a \equiv M \Delta u \Delta t$  (8)

where  $d_p$  is the particle diameter and a is its local displacement, M is the image magnification,  $\Delta t$  is the delay between two laser pulses and  $\Delta u$  the local variation of the velocity. In the case of the Diesel spray, the gradients of interest are the radial ones since they are higher than the axial ones. Using the 1d spray model, the radial gradients of axial velocity can be computed for square interrogation windows of 8, 16 and 32 pixels, at each axial location of the spray. Then the criteria defined in Eq. 8 can be estimated with M=9.4pxl/mm and  $\Delta t$ =4µs, and  $d_p$ =1pxl. The results are presented in Figure 9, showing that for

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an interrogation window of 8x8 pixels (resp. 16x16 and 32x32), the criterion is met when x>20mm (resp. x>30mm and x>40mm). Interestingly, the axial locations where the criteria are true are similar to the ones where the interrogation window size has no effect on velocity profile, as described in Figure 8.



Figure 8 - Influence of window size on the measured centerline axial velocity



Figure 9 - Axial evolution of |a| for different sizes of interrogation window, and comparison with  $d_n$  (blue dotted line)

However, even if an 8x8 window size captures a wider velocity range, the oscillations visible for the corresponding curve also shows that this lower window size also generates a lower measurement precision. As a compromise, in the present work the processing has been performed with 16x16 pixels interrogation windows. Figure 9, shows that at distances from the injector below 30mm the measurement are probably less reliable.

This analysis of the limits of the velocity measurements shows that a good accuracy is difficult to obtain for measurements at distances shorter than 30mm from the nozzle because of the inertia of the particles, the effect of time delay between laser pulses and/or of the interrogation window size. Therefore, in the following sections, only velocities downstream of 30mm will be considered. Besides, the inertia of the particles can also affect the velocities in the far field (x>30mm), so the results must be carefully analyzed. As discussed in

the introduction of this paper, these experimental results will be analyzed to validate the hypothesis of the 1D spray model, and to assess the consistency of the velocity and mixture fraction fields database.

#### Performances of the 1D spray model

Figure 10 and Figure 11 show the comparison between experimental and modeled data for each condition, for the axial component of the velocity and the mixture fraction, both on the spray axis. The solid lines are for the experimental results, the dashed lines are for the 1D spray model. As discussed earlier, the velocity fields are not reliable below 30mm, which is why the axial plots for velocity start at 30mm, compared to 20mm for the mixture fraction. Each color corresponds to one type of boundary conditions.



Figure 10 - Velocity (top) and mixture fraction (bottom) axial profiles for different boundary conditions. Solid lines represent the experimental results, dotted lines the model. The grey error bars represent the 95% confidence interval uncertainties

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# Figure 11 - Velocity (top) and mixture fraction (bottom) radial profiles for different boundary conditions at x=30mm. Solid lines represent the experimental results, black dotted lines the model.

Figure 10 and Figure 11 show good agreement between experiment and model for all the cases considered, within the uncertainty of the measurement, suggesting that the 1D spray model responds appropriately to the variations in the boundary conditions. Therefore, this result further validates the 1D model and its founding hypotheses. Since the model is based on the solving of mass momentum and mass conservation equations, it shows that the mixing process and flow dynamics of the spray are driven by the mass flow rate and mass momentum exchanges.

Besides, these figures show that the effects of boundary conditions variations on the velocity and mixture fraction fields are consistent with previous results from the literature [42–44]. A decrease of the injection pressure does not affect the mixture fraction field in the spray while the axial velocity is affected. This result is explained by the fact that although the mass momentum decreases, it has no effect on the mixture fraction field because the air entrainment is proportional to the injected fuel mass flow rate. A decrease of

ambient density affects both the mixture fraction and the velocity fields, which are increasing.

These results also allow the analysis of the ability of the ceramic particles used in the PIV technique to follow the actual flow. Indeed, it has been shown that:

- The experimental axial velocity profiles are very similar to the modelled ones, and that the model captures very well the behaviour of the real spray;
- The normalized evolution of axial mixture fraction and velocity profiles are also similar. Also, no major deviation due to particle inertia is observed.

These two observations tend to prove that the particles used in this study to measure the velocities in the spray are good markers of the flow, i.e. that they have a low enough inertia.

#### Consistency between PIV and Rayleigh measurements

In the previous section, velocity and mixture fraction fields (carried out with different ECN injectors) were compared to 1D model simulation results adapted to each injector characteristics. The velocity field was compared to model results using input momentum and mass flow rates and spray angle of injector 210678-aged, while mixture fraction fields were compared to model results using injector 210677 inputs. All these characteristics are listed in Table 1. The results presented in Figure 10 show that the results fall within the measurement uncertainty. But this analysis does not give a clue whether or not the velocity field obtained with injector 210678-aged is consistent with the mixture fraction field obtained with injector 210677. In other words, if the mixture fraction field had been measured on injector 210678-aged, would we have obtained the same results as obtained with injector 210677? This question is the core of the ECN philosophy. Indeed, the aim of the ECN is to build a unique and complete database, and this effort relies on the use of nominally identical injectors. Previous work has shown that in practice slight differences exist between these injectors [45]. It is therefore interesting to evaluate the effect of these differences on the consistency between the datasets. This question can be addressed by the use of the 1D spray model, which has been shown in the previous section to provide an accurate estimation of the behaviour of the real spray. Using this model, it is possible to estimate the mixture fraction field corresponding to injector 210678-aged, for which only the velocity field has been measured. And reciprocally, the velocity field of injector 210677 can also be estimated with the 1D spray model.

Figure 12 and Figure 13 compare experimental velocity (in gray) and mixture fraction axial profiles obtained with 1D model simulations corresponding to both injectors (colored lines, solid lines for injector 210677, dashed lines for injector 210678-aged). Regarding the mixture fraction fields, the modelling results for injector 210678aged are slightly higher than for injector 210677. However, the results (dashed line) are also within the uncertainty of the measurements, but the agreement is better for injector 210677, the one that was actually used for these mixture fraction measurements. Concerning the velocity fields, injector 210677 shows higher axial velocities than injector 210678-aged, in line with its higher momentum rating (Table 1). But once again, the actual injector used for the velocity study was 210678-aged. This therefore shows that although the ECN injectors are nominally identical, in practice existing differences have to be considered when comparing datasets obtained from different injectors. Of course, ideally injectors could

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be made to tighter specification, or aging could be controlled such that their performance would not change throughout time. These differences between nominally identical injectors have already been observed in [32, 45] using standard optical diagnostics for the characterization of spray penetration and combustion, but have not been demonstrated for velocity and mixture fraction requiring more complex diagnostics. Thus, the experimental database for velocity and mixture fraction fields, even if they are close, are not completely consistent with one another. The difference in the injected momentum between injectors 210677 and 210678-aged leads to significant differences in mixture fraction and velocity fields. But the 1D spray model, once calibrated, can be used to bring consistency between different experimental databases.

#### Conclusion

The process of Diesel spray mixture formation was investigated within the ECN framework. High-speed Particle Image Velocimetry was used to measure the velocity field inside the jet. The PIV measurements were carried out at Spray A target conditions with injector 210678-aged, enabling direct comparisons with mixture fraction measurements previously performed under the same conditions with injector 210677. A 1D spray model, based on fundamental hypothesis on the dominant processes of spray formation, was used to evaluate the latter hypothesis and to compare quantitatively the consistency between the two databases. The spray angle is fitted to match the measured vapor penetration. With this input vapor angle, the 1D spray model is able to predict quantitatively the main spray characteristics (average mixture fraction and velocity fields) within the measurement uncertainties and for a wide range of parametric variations. At last, the 1D spray model has been used to show that the differences between injector 210677 and 210678-aged will induce slightly different velocity and mixture fraction fields. However, a consistent database can be built using advanced diagnostics performed by different institutions when the boundary conditions are well known like in the ECN Spray A framework, and when the 1D model is used to assess the difference between injectors characteristics. In a general perspective, these results show that the global processes of mixture formation are well understood, but that in order to enable fully predictive simulations, the key is the understanding of the physical processes that determine the angle of the spray. This will only be achieved when the relation between internal nozzle flow and near field spray structure is well understood. This is a major challenge that has been identified within the ECN collaboration. Also improvements of the measurement accuracy in the high velocity regions close to the liquid spray tip should be considered.



Figure 12 - Comparison of mixture fractions experimental results (in gray) with modeled ones. Solid lines are the estimations from the 1D spray model for injector 210677; dashed lines for injector 210678-aged.



Figure 13 - Comparison of velocity experimental results (in gray) with modeled ones. Solid lines are the estimations from the 1D spray model for injector 210677; dashed lines for injector 210678-aged.

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