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# An Updated Empirical Correlation Formalism for Laminar Flame Speeds: Application to a TRFE Gasoline Surrogate in Highly Diluted Conditions

Boyang Xu<sup>a,b,\*</sup>, Pascal Diévar<sup>b</sup>, Mickaël Matrat<sup>a</sup>, Julian Garrec<sup>b</sup>, Laurent Catoire<sup>b</sup>

<sup>a</sup>*IFP Energies nouvelles, 1 et 4 avenue de Bois-Préau, 92852 Rueil-Malmaison, France ; Institut Carnot IFPEN Transports Energie*

<sup>b</sup>*Unité Chimie et Procédés, ENSTA Paris, Institut Polytechnique de Paris, 91120 Palaiseau, France*

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## Abstract

In order to comply with present and future stringent environmental policies, engine manufacturers have to improve engine design and control to achieve combustion with high dilution ratios. The use of CFD simulations with complex combustion chemistry remains prohibitive, and alternatives to assess quickly the laminar flame speeds at local grid cell conditions (temperature, pressure, equivalence ratio, and dilution ratio) are desired, such as empirical correlations. Whereas these correlations should ideally be obtained from experimental measurements, comprehensively validated kinetic models can help extend databases to conditions that cannot be achieved practically. Hence, the present study proposed a reduced kinetic mechanism, containing 593 species and 3698 reactions, for one gasoline surrogate, namely Toluene Reference Fuel with Ethanol addition (TRFE). It is obtained by first compiling sub-mechanisms from the literature for the four components (isooctane, n-heptane, toluene, and ethanol) with updates of some key rate constants, and then is reduced for 1-D flame speed computations. The model was first validated against recent experimental laminar flame speed measurements of TRFE/air/diluent mixtures for various temperatures, pressures, equivalence ratios, and dilution ratios and then employed to extend the experimental database for the TRFE surrogate to higher dilution ratios and temperatures. A new formalism, including new mathematical expressions for the reference and dilution terms, and the temperature exponent are proposed. This new formalism exhibits improved abilities in fitting the laminar flame speeds, especially at high dilution ratios and in very fuel-lean

and fuel-rich conditions. A new mathematical correlation based on these formulas was developed, whose correlation parameters were obtained by fitting both experimental data and mechanism predictions. Regardless of the conditions, the present correlation is observed to show overall good agreements with available experimental data in the literature for laminar flame speeds and their dependence on equivalence ratio, temperature, pressure, and dilution. Results show that the effect of dilution in reducing flame speeds is not linear and depends on the equivalence ratio. A parameter  $\mu$  is defined in the correlation formulas to evaluate the dilution effectiveness of the diluent, which is found to be composition-specific and possibly follows a linear mixing rule for diluent mixtures.

*Keywords:*

laminar flame speed, gasoline surrogate, empirical correlation, dilution, exhaust gas recirculation, kinetic mechanism

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\*Corresponding author

*Email addresses:* [boyang.xu@ifpen.fr](mailto:boyang.xu@ifpen.fr) (Boyang Xu), [boyang.xu@outlook.com](mailto:boyang.xu@outlook.com) (Boyang Xu)

## 1 Nomenclature

2	$\alpha$	Temperature exponent
3	$\beta$	Pressure exponent
4	$\gamma$	Oxygen ratio exponent
5	$\mu$	Dilution effectiveness
6	$\mu_\alpha$	Temperature exponent for dilution effectiveness
7	$\mu_\beta$	Pressure exponent for dilution effectiveness
8	$\mu_i$	Dilution effectiveness for individual components in the diluent mixture
9	$\mu_{mix}$	Dilution effectiveness for the diluent mixture
10	$\nu_{O_2}$	Oxygen ratio in the non-fuel fraction of the mixture
11	$\rho$	Density
12	$v$	liquid volume fraction of ethanol in fuel
13	$\varphi$	Equivalence ratio
14	$\varphi_\alpha$	Median equivalence ratio for temperature exponent
15	$\varphi_\beta$	Median equivalence ratio for pressure exponent
16	$\varphi_\mu$	Median equivalence ratio for dilution effectiveness
17	$\varphi_m$	Median equivalence ratio
18	$n_{O_2}$	amount of oxygen molecules in moles
19	$P$	Pressure
20	$P_i$	Initial pressure

- 21  $P_{ref}$  Reference pressure
- 22  $R^2$  Coefficient of Determination
- 23  $S_{L,max}$  Maximum laminar flame speed
- 24  $S_{L,ref}$  Reference laminar flame speed
- 25  $T$  Temperature
- 26  $T_i$  Initial temperature
- 27  $T_{ref}$  Reference temperature
- 28  $X_{air}$  Mole fraction of air in the mixture
- 29  $X_{EGR}$  Mole fraction of EGR in the mixture
- 30  $X_{fuel}$  Mole fraction of fuel in the mixture
- 31  $x_d$  Dilution ratio
- 32  $X_i$  Mole fraction
- 33  $X_{et}$  Mole fraction of ethanol in fuel
- 34  $S_L$  Laminar flame speed
- 35 CFD Computational Fluid Dynamics
- 36 CURV Adaptive Grid Control Based On Solution Curvature
- 37 DRGEP Direct Relation Graph method with Error Propagation
- 38 EGR Exhaust Gas Recirculation
- 39 FSSA Full Species Sensitivity Analysis
- 40 GRAD Adaptive Grid Control Based On Solution Gradient

- 41 HCs Unburnt Hydrocarbons
- 42 MON Motor Octane Number
- 43 NOx Nitrogen Oxides
- 44 p.w. Present Work
- 45 RMSE Root Mean Squared Error
- 46 RON Research Octane Number
- 47 SA Sensitivity Analysis
- 48 SI Spark Ignition
- 49 TRF Primary Reference Fuel
- 50 TRF Toluene Reference Fuel
- 51 TRFE Toluene Reference Fuel with Ethanol addition

## 52 1. Introduction

53 Engine manufacturers have to comply with more and more stringent environmental public  
54 policies, thus pressing to reduce both pollutants emissions (CO, NOx, HCs) and fuel con-  
55 sumption. Over the last decades, Exhaust Gas Recirculation (EGR), as a mean to achieve  
56 diluted combustion, has been a technology that benefited Spark Ignition (SI) engines since  
57 it has been demonstrated to reduce knock tendency, mitigate NOx emission while promoting  
58 fuel savings. [1–5] However, engine operation in highly-diluted regimes (EGR ratio greater  
59 than 25%) remains difficult mainly because of the slower heat release rate and the deterio-  
60 rated engine stability induced. [6] These issues could not be overcome by the measures that  
61 proved efficient at lower dilution ratios such as increasing turbulence level and advanced  
62 spark-ignition technologies (including increasing ignition energy). [6] In order to propose and  
63 develop new strategies to control the combustion at these higher dilution rates, extensive use  
64 of turbulent Computational Flow Dynamic (CFD) is required. These CFD codes are based  
65 on the flamelets regime assumption, in which the laminar flame speed ( $S_L$ ) at the grid-cell  
66 local conditions (temperature, pressure, equivalence ratio, dilution ratio) is the fundamental  
67 property to be determined, explaining the reason why laminar flame speeds measurements  
68 are still of great interest. [7] Although kinetic models can be embedded into CFD codes  
69 to determine on-the-fly  $S_L$ , this approach is limited to small mechanisms (usually less than  
70 50 species) which are challenging to develop with sufficient accuracy for complex fuels. [8]  
71 Hence, other strategies have been devised such as empirical correlations.

72 The commonly used mathematical correlation relates the laminar flame speeds to the  
73 equivalence ratio, temperature and pressure,  $S_L = f(\varphi, T, P)$ .

$$S_L = S_{L,ref}(\varphi) \left( \frac{T}{T_{ref}} \right)^\alpha \left( \frac{P}{P_{ref}} \right)^\beta \quad (1)$$

74 where  $S_{L,ref}(\varphi)$ , the flame speed at a reference condition ( $T_{ref}, P_{ref}$ ), is a quadratic function  
75 of the equivalence ratio, and the exponent  $\alpha$  and  $\beta$  are usually linear functions of the  
76 equivalence ratio.

77 With the development of EGR technologies, the necessity to consider the effect of the  
78 EGR dilution ratio ( $x_d$ ) and composition led to the introduction of a corrective term  $f(x_d)$ .

79 Early studies by Ryan and Lestz [9], Metghalchi and Keck [10], Gülder [11, 12] reported  
80 laminar flame speeds of both single-component fuels and commercial gasoline at various  
81 temperatures, pressures, and dilution ratios. The three studies used a synthetic N<sub>2</sub>/CO<sub>2</sub>  
82 blend (85%/15% by volume, with small variations between the three studies) as the diluent  
83 since it has heat capacities similar to the exhaust gas. They observed that, within their  
84 investigation range (up to 30% dilution),  $S_L$  decreases linearly with the dilution ratio and  
85 the decay rate was independent of the unburnt gas temperature. Kumar *et al.* [13] drew  
86 the same conclusion by investigating isooctane/air and n-heptane/air flames diluted in N<sub>2</sub>.  
87 Therefore, the first attempts [9–12] to model the dilution factor adopted a linear expression,

$$f_1(x_d) = 1 - ax_d \quad (2)$$

88 in which the parameter  $a$  was found to lie within 2.1 and 2.5.

89 Rhodes and Keck [14] measured the propagation speed of spherically expanding flames  
90 of two multicomponent gasoline surrogates, with and without prior dilution in a synthetic  
91 EGR (N<sub>2</sub>/CO<sub>2</sub> = 80%/20% by volume). Unlike previous studies, the dilution effect was  
92 observed to be not linear, and the authors thus introduced an exponent on the dilution ratio

$$f_2(x_d) = 1 - ax_d^b \quad (3)$$

93 and determined  $a$  and  $b$  to be equal to 2.06 and 0.77, respectively, for both fuels investigated.

94 As an alternative to experimental measurements, Syed *et al.* [15] carried out a modeling  
95 study in which the effect of ethanol blending and EGR dilution on the propagation of  
96 gasoline/air flames was investigated. In line with the conclusions of Rhodes and Keck,  
97 the simulated data revealed a non-linear dependence on the dilution ratio, which was best  
98 captured by the following expression

$$f_3(x_d) = (1 - ax_d)^b \quad (4)$$

99 where the two coefficients  $a$  and  $b$  vary slightly with the ethanol fraction.

100 Bhattacharya *et al.* [16] measured using the heat-flux burner technique the atmospheric  
101 laminar flame speeds of gasoline/air mixtures diluted in up to 15% of N<sub>2</sub>. They observed that

102 the laminar flame speed decrease caused by the N<sub>2</sub> dilution was equivalence ratio dependent,  
 103 and thus adopted for the dilution factor an expression similar to the one proposed by Rhodes  
 104 and Keck (Eq. 3), but in which the exponent is a linear function of the equivalence ratio.

$$f_2(x_d) = 1 - \mu_1 x_d^{\mu_2 + (\varphi - 1)\mu_3} \quad (5)$$

105 This expression was initially suggested by Clarke [17, 18] in a study on the propagation of  
 106 fuel/air flames in zero gravity conditions. A few years later, Marshall *et al.* [19], in an ex-  
 107 tensive study on the laminar flame speeds of several neat liquid fuels (e.g., n-heptane, isooc-  
 108 tane, toluene, ethylbenzene, and ethanol) diluted in real EGR obtained from preliminary  
 109 explosions, employed a similar expression to quantify the dilution effect. The parametriza-  
 110 tion based on the experimental data showed that the values of the three parameters were  
 111 strongly fuel dependent.

112 In their study on the laminar flame speeds of isooctane, n-butanol, and their blends, Fu  
 113 *et al.* [20] expanded the original expression of Clarke by introducing a quadratic dependence  
 114 of the exponent for the equivalence ratio to recover better the dependence of the dilution  
 115 factor. Moreover, they offered a simple mixing rule able to calculate the dilution factor of a  
 116 multicomponent EGR from the parameters of the neat diluents,

$$f_2(x_d) = 1 - \sum_{i=1}^n X_i \mu_{1,i} x_d^{\mu_{2,i} + \mu_{3,i}(\varphi - \varphi_{m,i}) + \mu_{4,i}(\varphi - \varphi_{m,i})^2} \quad (6)$$

117 where  $\varphi_{m,i}$  is an arbitrarily-set median equivalence ratio for the  $i$ th component in the EGR  
 118 mixture. The authors also questioned the dependence of the dilution factor on the initial  
 119 temperature and pressure, and concluded from kinetic modeling results that such dependence  
 120 could be safely neglected.

121 Halter and co-workers [21–24] published a series of papers quantifying the effect of sev-  
 122 eral diluents and additives on the propagation speed of isooctane/air flames. The diluent  
 123 considered were N<sub>2</sub>, CO<sub>2</sub>, H<sub>2</sub>O, and a ternary synthetic exhaust gas (N<sub>2</sub> / CO<sub>2</sub> / H<sub>2</sub>O). The  
 124 experimental observations indicate that diluent inhibition effectiveness decreases in the or-  
 125 der: CO<sub>2</sub>, H<sub>2</sub>O, synthetic EGR, and N<sub>2</sub>. The large collection of data was also used to  
 126 propose an alternative to the traditional dilution factor expression by introducing the ratio,

127  $\nu_{O_2}/\nu_{O_2,ref}$ , where  $\nu_{O_2}$  is the ratio of oxygen in the non-fuel fraction of the mixture, i.e.,  
 128  $X_{O_2}/(X_{air} + X_{diluent})$ , and  $\nu_{O_2,ref}$  is the same ratio but for the reference mixture, which is  
 129 the mixture without dilution. The dilution factor is then expressed as

$$f_4(\nu_{O_2}) = \left( \frac{\nu_{O_2}}{\nu_{O_2,ref}} \right)^\gamma \quad (7)$$

130 where the exponent  $\gamma$  is a linear function of the equivalence ratio [24]. The parametriza-  
 131 tion was only conducted for N<sub>2</sub> dilution, and  $\gamma$  was found to be weakly dependent on the  
 132 equivalence ratio for isooctane ( $d\gamma/d\varphi = 0.17$  and  $\gamma \approx 2.65$ ).

133 More recently, Di Lorenzo *et al.* [25] measured the laminar flame speeds of a gasoline and  
 134 its TRFE surrogate in air, either diluted (up to 20%) or not by synthetic EGR (13.62% CO<sub>2</sub>,  
 135 12.22% H<sub>2</sub>O and 74.16% N<sub>2</sub>). Alongside the observation that the proposed TRFE surrogate  
 136 was suitable to mimic the flame behavior of the real gasoline, even at diluted conditions,  
 137 they successfully applied the dilution factor expression proposed by Galmiche *et al.* [23]. In  
 138 contrast with the observations for N<sub>2</sub> diluted isooctane/air flames, a stronger dependence of  
 139 the exponent  $\gamma$  to the equivalence ratio was reported for both the gasoline and its surrogate  
 140 ( $\gamma = 2.95 - 0.43\varphi$ ).

141 To sum up, several expressions for the dilution factor have been proposed (as listed in  
 142 Table 1). These expressions got more complex as non-linear and equivalence ratio dependen-  
 143 cies were introduced. Up to now, these modifications have always been proposed based on  
 144 experimental observations and with the aim of retaining simple mathematical expressions.

145 Besides empirical correlations, Cho and Song [26] recently investigated the dilution ef-  
 146 fect on the ignition delay times from a theoretical perspective. They distinguished the  
 147 non-chemical and chemical contributions. They related the former to the thermochemical  
 148 properties of the diluent and the fuel/air mixtures, namely the ratio of constant volume  
 149 heat capacities and the average Zeldovich number, whereas the latter is associated with  
 150 the change in the overall mixture collision efficiency. They finally proposed a mathematical  
 151 correlation suited for ignition delay times prediction. Although the conclusions of Cho and  
 152 Song are only applicable to 0D-ignition, they could be the ground for a similar approach for  
 153 flame propagation.

Table 1: Mathematical expressions from the literature modeling the dilution effect on the laminar flame speeds.  $x_d$  is the dilution ratio. Other parameters in the equations are described in the text.

No.	Equation	Fuel	Diluent	Max $x_d$	T (K)	P (bar)	$\varphi$	$S_L$ database a	Ref.
1	$f(x_d) = 1 - ax_d$	isooctane, n-heptane, methanol, ethanol, methane, propane, IRMFD-303 (indolene)	N <sub>2</sub> /CO <sub>2</sub> (85/15)	30%	300– 700	0.4– 50	0.7– 1.4	Exp.	[9– 12]
2	$f(x_d) = 1 - ax_d^b$	IRMFD-303 (indolene), RMFD-302	N <sub>2</sub> /CO <sub>2</sub> (80/20)	30%	350– 550	0.4– 12	0.7– 1.6	Exp.	[14]
3	$f(x_d) = (1 - ax_d)^b$	TRFE <sup>b</sup>	CO <sub>2</sub> /H <sub>2</sub> O/N <sub>2</sub>	32%	300– 600	4–8	1.0	Mod.	[15]
4	$f(x_d) = 1 - \mu_1 x_d^{\mu_2 + (\varphi - 1)\mu_3}$	commercial gasoline, n-heptane, isooctane, toluene, ethylbenzene, ethanol	N <sub>2</sub> , real EGR	30%	310– 450	0.5–4	0.7– 1.3	Exp.	[16– 19]
5	$f(x_d) = 1 - \sum_{i=1}^n X_i \mu_{1,i} x_d^{\mu_i}$ $\mu = \mu_{2,i} + \mu_{3,i}(\varphi - \varphi_{m,i}) + \mu_{4,i}(\varphi - \varphi_{m,i})^2$	isooctane, n-butanol	CO <sub>2</sub> , H <sub>2</sub> O, N <sub>2</sub> , CO <sub>2</sub> /H <sub>2</sub> O/N <sub>2</sub>	30%	428– 500	1–5	0.7– 1.5	Mod.	[20]
6	$f(\nu_{O_2}) = \left( \frac{\nu_{O_2}}{\nu_{O_2,ref}} \right)^\gamma$	isooctane, TRFE <sup>b</sup>	CO <sub>2</sub> , H <sub>2</sub> O, N <sub>2</sub> , CO <sub>2</sub> /H <sub>2</sub> O/N <sub>2</sub>	25%	373– 473	1–10	0.8– 1.5	Exp.	[21– 25]

<sup>a</sup> Expt.= experimental data, Mod. = Kinetic Modeling.

<sup>b</sup> TRFE: Toluene Reference Fuel with Ethanol Addition, a gasoline surrogate composed of n-heptane, isooctane, toluene, and ethanol.

154 Nevertheless, some authors have already attempted to assess the non-chemical and chem-  
155 ical contribution of the dilution dumping effect on the laminar flame speed by using detailed  
156 chemical kinetic models. By considering fictitious species retaining only some properties of  
157 the diluent, the non-chemical and chemical contributions can be uncoupled. Such analyses  
158 demonstrated that the dilution effect of N<sub>2</sub> was almost exclusively dominated by its thermal  
159 properties, whereas thermal and chemical mechanisms were both important for CO<sub>2</sub> and  
160 H<sub>2</sub>O. [21, 27, 28] The chemical mechanisms involved appear to be fuel independent, as they  
161 proceed through the interactions with the radical pool. Regarding CO<sub>2</sub>, these interactions  
162 mainly consist in its reaction with the H atom (CO<sub>2</sub> + H = CO + OH) and its larger Chap-

163 eron efficiency promoting termination reactions such as  $\text{H} + \text{O}_2 (+ \text{M}) = \text{HO}_2 (+ \text{M})$  and  
164  $\text{H} + \text{OH} (+ \text{M}) = \text{H}_2\text{O} (+ \text{M})$ , and to a lesser extent with its reactions with  $\text{CH}_3$ ,  $\text{CH}_2$   
165 and  $\text{CH}$  radicals. [28] For water, the chemical interactions remain less clear: besides its  
166 participation as an efficient third body collider in termination reactions, its participation in  
167 a large body of H-abstraction reactions may also displace some chemical equilibria. [29]

168 However, the aforementioned correlations show limitations as the interest is growing  
169 for combustion at more extreme conditions, such as over-lean and over-rich mixtures, high  
170 temperature, high pressure, and high dilution ratios. For example, the  $S_L$  vs.  $\varphi$  curve  
171 deviates from the quadratic shape as the equivalence ratio expands to over-lean and over-  
172 rich conditions. Indeed, it is asymmetric between the left (or “lean”) and right (or “rich”)  
173 branches:  $S_L$  decreases to an asymptotic value as  $\varphi$  approaches the flammability limits [30],  
174 unlike the faster decay predicted by the quadratic expression which would result in large  
175 uncertainties near the flammability limits. A similar limitation also holds for the temperature  
176 exponent [30], with only the convexity of the parabola being the opposite. Regarding the  
177 effect of dilution, an increasing number of studies suggest a non-linear dependence of  $S_L$   
178 with the dilution ratio, but existing expressions show either inadequate fitting quality, large  
179 uncertainties, or unrealistic values at high dilution ratios. Deficiencies of these expressions  
180 originate from the lack or the limited set of experimental data at these unusual conditions.  
181 However, this situation can nowadays be fixed through kinetic modeling.

182 The aim of the present work is to propose new expressions for the laminar flame speed  $S_L$   
183 correlation, especially for the reference  $S_L$ , temperature exponent, and dilution terms. These  
184 new formulations will then be applied to a Toluene Reference Fuel with Ethanol addition  
185 (TRFE) emulating a commercial gasoline fuel for which an experimental  $S_L$  database is  
186 available. This latter will then be extended through kinetic modeling, employing a kinetic  
187 mechanism specifically herein developed and validated for this surrogate. Finally, correlation  
188 parameters will be derived and the adequacy of the proposed empirical correlation will be  
189 discussed.

## 190 2. Empirical Correlation

191 As mentioned in the introduction, empirical correlations enabling quick estimation of the  
192 laminar flame speeds for a given set of conditions (T, P, equivalence ratio  $\varphi$  and dilution  
193 ratio  $x_d$ ) are widely used in CFD modeling. In order to expand the validity of correlation  
194 to wider ranges of equivalence ratio, temperature, pressure, and dilution ratio, we propose  
195 a new correlation with novel expressions. The correlation formula takes the form expressed  
196 as Eq. 8, with addition of a correction term for dilution with respect to Eq. 1.

$$S_L(\varphi, T, P, x_d) = S_{L,ref}(\varphi) \left( \frac{T}{T_{ref}} \right)^\alpha \left( \frac{P}{P_{ref}} \right)^\beta \exp \left( - \frac{\mu x_d}{1 - x_d} \right) \quad (8)$$

197 The details of each term in Eq. 8 are discussed in the following text of this section. It is  
198 noteworthy that the expressions discussed in this section are universal and apply to any fuel.

### 199 2.1. Reference Laminar Flame Speed

200 Since the pioneering work of Metghalchi *et al.* [31, 10], polynomial functions, espe-  
201 cially second-order, have been preferentially used to fit laminar flame speeds. Indeed, both  
202 experimental measurements and modeling results show that the laminar flame speed depen-  
203 dence with the equivalence ratio exhibits a symmetry axis around  $\varphi = 1.05$ – $1.10$  for most  
204 of the fuels. Nevertheless, this observation does not hold for the very-lean and very-rich  
205 fuel branches, and the quadratic expression may then underestimate strongly the laminar  
206 flame speed, and even returns negative values. Some authors have circumvented this issue  
207 by adopting quartic expressions. [23] But, as shown thereafter, such expressions still under-  
208 estimate the laminar flame speeds in very fuel-rich conditions while introducing new issues.  
209 Therefore, a different formalism is desired.

210 The behavior of the laminar flame speed on the fuel-lean and fuel-rich side is very dif-  
211 ferent, and therefore each branch should be described by its own representative function.  
212 Moreover, the proposed expression should also be able to capture the first derivative of the  
213 laminar flame speed with respect to the equivalence ratio, especially the dampening usu-  
214 ally observed in over-rich conditions [30], enabling a more realistic representation near the  
215 flammability limits. And ideally, the parameters involved in the expression would be related

216 to physical quantities. Within these considerations, we propose the following correlation in  
 217 which Gaussian functions are preferred over polynomial expansions.

$$S_{L,ref}(\varphi) = \begin{cases} A_1 \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_m}{B_1} \right)^2 \right] + (S_{L,max} - A_1), & \varphi > \varphi_m \\ A_2 \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_m}{B_2} \right)^2 \right] + (S_{L,max} - A_2), & \varphi \leq \varphi_m \end{cases} \quad (9)$$

218 The new expression herein proposed requires six parameters, and they are associated with  
 219 physical quantities as demonstrated in Figure 1a and 1b.  $S_{L,max}$  is the maximum value of  
 220 the laminar flame speed that is observed at the equivalence ratio  $\varphi_m$ . The parameter  $A_1$   
 221 correspond to the difference between the  $S_{L,max}$  and the laminar flame speed at the fuel-rich  
 222 flammability limit. It is worth noting that although it should be the same for  $A_2$  (in the fuel-  
 223 lean branch) based on Eq. 9, fitting based on the limited data available would yield an  $A_2$   
 224 value larger than  $S_{L,max}$  and therefore negative  $S_L$  as  $\varphi$  approaches to zero. It indicates that  
 225 a better and more realistic expression for the fuel-lean branch is still needed and subjects  
 226 to further studies. The constants  $B_1$  and  $B_2$  are the equivalence ratio differences between  
 227 the central value  $\varphi_m$  and the mixtures of maximum rates of  $S_L$  decrease and increase on the  
 228 fuel-rich and fuel-lean branches, respectively, i.e.,  $\left( \frac{d^2 S_L}{d\varphi^2} \right)_{\varphi_m+B_1} = \left( \frac{d^2 S_L}{d\varphi^2} \right)_{\varphi_m-B_2} = 0$ .

229 Figures 1a and 1b compare the fitting quality of the present correlation with the con-  
 230 ventional quadratic expression as well as the quartic expression regarding the laminar flame  
 231 speed and its derivative with respect to the equivalence ratio. A better agreement is observed  
 232 on the whole equivalence ratio domain, especially at the edges. In contrast, the quartic ex-  
 233 pression exhibits an unrealistic nonmonotonic behavior at the over-rich edges, indicating its  
 234 insufficiency for fitting laminar flame speeds at extreme equivalence ratios.

235 Moreover, the fitting capability of Eq. 9 is further validated based on the experimental  
 236 flame speeds for  $\text{CH}_4$ ,  $\text{C}_2\text{H}_6$ , and  $\text{C}_3\text{H}_8$  at over-rich conditions measured by Han *et al.* [30],  
 237 as shown in Figure 2. For over-lean conditions, although the bending of the curve is usually  
 238 less significant, the Gaussian function is also adopted which could be helpful when new  
 239 experimental data become available. These indicate that the use of Gaussian functions  
 240 could better represent flame speeds for extreme (over-lean and over-rich) mixtures, and  
 241 could help expand the validity of correlations in terms of equivalence ratio.

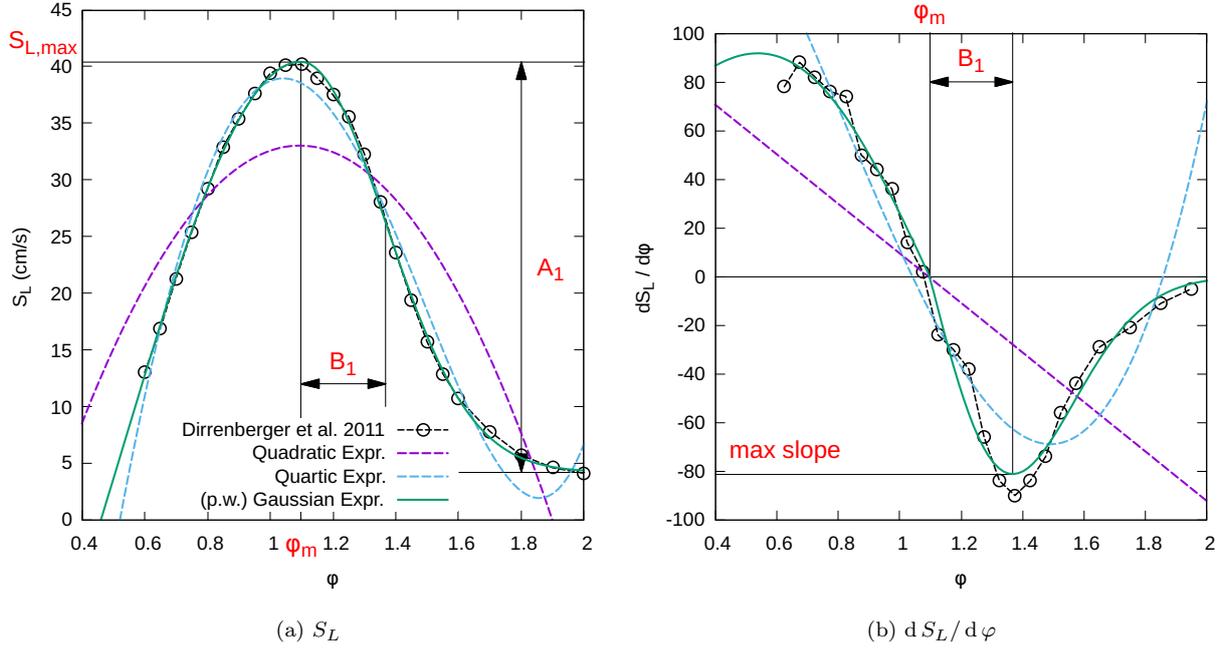


Figure 1: (a) Laminar flame speed and (b) its derivative  $dS_{L,ref}/d\phi$  as a function of equivalence ratio: comparison between the herein proposed Gaussian, the usual quadratic, and the more recent quartic expressions. The experimental data used as reference are propane/air flames at standard conditions (300 K, 1 atm) from Dirrenberger *et al.* [32].

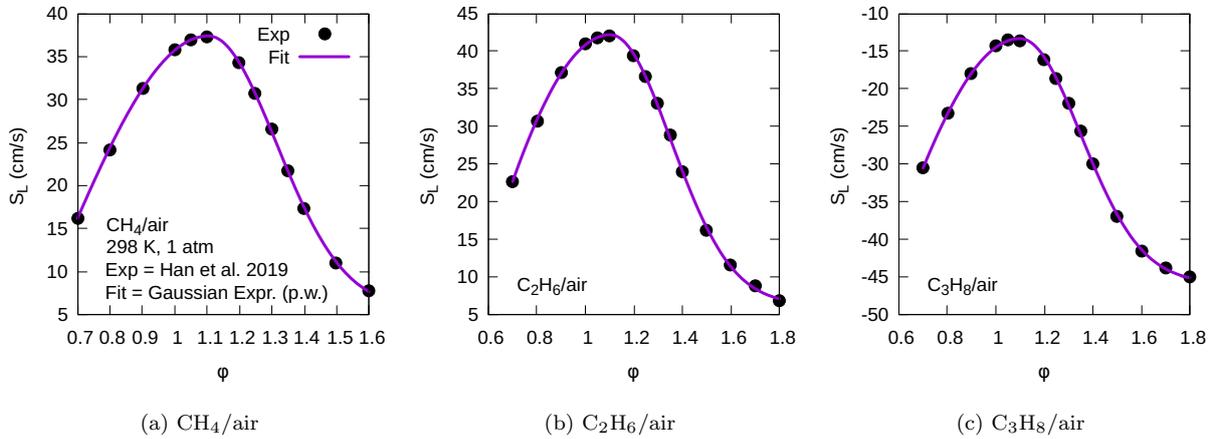


Figure 2: Laminar flame speed for (a)  $CH_4$ , (b)  $C_2H_6$ , and (c)  $C_3H_8$ : comparison between experimental data [30] and the herein proposed Gaussian expression.

## 242 2.2. Temperature Exponent

243 Similar to the reference flame speed correlation, quadratic formulas are widely used in  
 244 the literature to model the temperature exponent. However, such functions result in a

245 temperature exponent that keeps increasing as the equivalence ratio increases or decreases,  
 246 resulting in unrealistic large values away from the stoichiometry. Recent experimental data  
 247 on C<sub>1</sub>–C<sub>3</sub> alkanes [30], on the contrary, show that for fuel-rich mixtures the temperature  
 248 exponent reaches an asymptotic value. Therefore, the step function consisting of two Gaus-  
 249 sian functions, similar to the reference flame speed term, is also necessary to avoid problems  
 250 of unrealistic (increasing)  $S_L$  at over-rich or over-lean conditions. With such considerations,  
 251 we propose the following equation for the temperature exponent.

$$\alpha(\varphi) = \begin{cases} \alpha_0 + \alpha_{A_1} - \alpha_{A_1} \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_\alpha}{\alpha_{B_1}} \right)^2 \right], & \varphi > \varphi_\alpha \\ \alpha_0 + \alpha_{A_2} - \alpha_{A_2} \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_\alpha}{\alpha_{B_2}} \right)^2 \right], & \varphi \leq \varphi_\alpha \end{cases} \quad (10)$$

252 Similar to Eq. 9,  $\alpha_0$  is the maximum or minimum value of the temperature exponent which  
 253 occurs at the equivalence ratio  $\varphi_\alpha$ .  $\alpha_{A_1}$ ,  $\alpha_{B_1}$ ,  $\alpha_{A_2}$ ,  $\alpha_{B_2}$  also have similar meanings on the  
 254 curves as the  $A_1$ ,  $B_1$ ,  $A_2$ , and  $B_2$  for  $S_{L,ref}$ , respectively.

255 Figure 3 shows the the comparison between the quadratic and Gaussian expression on  
 256 fitting the laminar flame speeds and their temperature exponent. The reference data used for  
 257 the fitting are for the TRFE fuel diluted by synthetic EGR, obtained by kinetic modeling  
 258 using MACDIL Mech.. Thier information is available in Section 3. It can be seen from  
 259 Figure 3a that the flame speeds generated using the conventional quadratic expression for  
 260 the temperature exponent exhibit an unrealistic increase at fuel-rich conditions ( $\varphi=1.8$ – $2.0$ ).  
 261 According to Figure 3b, this is due to the rapid increase of temperature exponent at rich  
 262 conditions (from  $\approx 3.4$  at  $\varphi = 1.8$  to  $\approx 4$  at  $\varphi = 2.0$ ), which is much faster than predicted  
 263 by MACDIL Mech.. In contrast with the quadratic expression, Eq. 10 results in a slower  
 264 increase of  $\alpha(\varphi)$  in the over fuel-rich region, and would eventually reach an asymptotic value.

265 In addition, the adequacy of Eq. 10 in modeling the temperature exponents is further  
 266 demonstrated against the experimental data for n-heptane and isooctane flames reported by  
 267 Han *et al.* [33] as displayed in the *Supplementary Material*.

268 It is noteworthy that although Eq. 10 is an improvement over the traditional quadratic  
 269 expression, it cannot eliminate completely the temperature exponent overestimation issue  
 270 at  $\varphi > 2$ , especially if neither experimental nor modeling data are available to constraint

271 the parameters optimization.

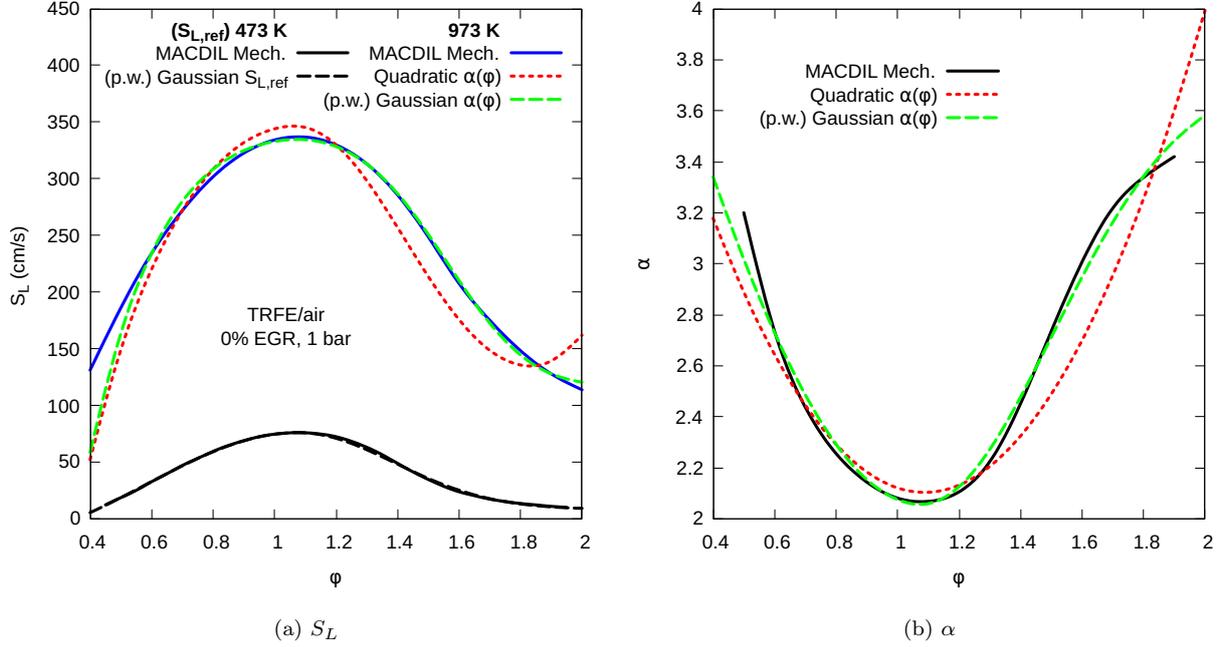


Figure 3: Comparison between (a) the laminar flame speeds (473 K and 973 K, 1 bar) and (b) the equivalence ratio dependent temperature exponent  $\alpha(\varphi)$  with either a quadratic or a Gaussian expression (Eq. 10) for the temperature exponent. The reference data used are for the TRFE fuel diluted by synthetic EGR, computed using MACDIL Mech.. (See Section 3)

### 272 2.3. Pressure Exponent

273 Since the early developments of the asymptotic theory of premixed flame, laminar flame  
 274 speed is known to decrease with the pressure ( $S_L \propto P^\beta$ ) with exponent  $\beta$  in the  $[-1, 0)$   
 275 interval. Such behavior has been experimentally confirmed, regardless of the fuel or the  
 276 equivalence ratio. Because of the complex chemistry involved in combustion, this exponent  
 277 changes with the mixture composition and therefore equivalence ratio dependent expressions  
 278 are required in the laminar flame speed correlations. Most of the recent studies adopted a  
 279 quadratic polynomial for  $\beta$ , and the present correlation stick with this usage,

$$\beta(\varphi) = \beta_0 + \beta_2(\varphi - \varphi_\beta)^2 \quad (11)$$

280 where  $\varphi_\beta$  is the equivalence ratio where the quadratic  $\beta$ - $\varphi$  curve is at its symmetrical axis.

281 *2.4. Dilution Factor*

282 Another originality of the present correlation is the formulation for the dilution factor  
 283  $f(x_d)$ . We propose an exponential expression

$$f(x_d) = \exp\left(-\frac{\mu x_d}{1 - x_d}\right) \quad (12)$$

284 which has the following mathematical characteristics to better match the evolution of  $S_L$   
 285 with the dilution ratio  $x_d$ : (i)  $f(0) = 1$ , i.e., the function is equal to unity at non-diluted  
 286 case ( $x_d = 0$ ); (ii)  $\lim_{x_d \rightarrow 1} f(x_d) = 0$ , i.e., the function approaches asymptotically to zero  
 287 as the dilution ratio keeps increasing; (iii)  $f'(x_d) < 0$ ,  $0 \leq x_d < 1$ , i.e., the function  
 288 decreases monotonically; (iv)  $\lim_{x_d \rightarrow 1} f'(x_d) = 0$ , i.e., the reduction effectiveness decreases  
 289 as the dilution ratio gets higher.

290 Because of the above characteristics, the proposed dilution term outperforms other exist-  
 291 ing formulas in the literature in capturing the dilution effect. Figure 4a compares calculated  
 292 laminar flame speeds for EGR dilution ratio up to 50% (higher dilution ratio could not be  
 293 reached because of numerical unstabilities due to the very low flame speeds) with the differ-  
 294 ent dilution terms discussed above. As it can be seen in this figure, expressions  $f_1(x_d)$  (Eq.  
 295 2) and  $f_2(x_d)$  (Eq. 3) generate unrealistic negative values at high dilution. The expression  
 296  $f_4(x_d)$  (Eq. 7) proposed in Ref. [25] could fit well at low dilution but deviates from data  
 297 at higher dilution, e.g.,  $x_d > 0.4$ . Although expression  $f_3(x_d)$  (Eq. 4) fits well the data, it  
 298 reaches zero at  $x_d = 1/a$  and yield erroneous values at  $x_d > 1/a$  (either negative or positive  
 299 depend on the value of parameter  $b$ ). Only Eq. 12 ( $f_0(x_d)$  in the figure) could fit well the  
 300 data and prevent yielding unrealistic values at all conditions at the same time.

301 In addition, the only parameter introduced in this expression,  $\mu$ , relates to a physical  
 302 property: it is the  $S_L$  reduction effectiveness of the diluent (or the diluent mixture) at zero-  
 303 dilution,  $\mu = -f'(0)$ . The higher the parameter  $\mu$  is, the stronger (more effective) the diluent  
 304 is in reducing flame speeds, as shown in Figure 4b. For example,  $\text{CO}_2$  is more efficient in  
 305 reducing  $S_L$  than  $\text{H}_2\text{O}$ , so the parameter  $\mu$  for  $\text{CO}_2$  is higher than that of  $\text{H}_2\text{O}$ . Therefore, the  
 306 formula proposed in the present work has the advantage of good fitting performance while  
 307 preventing unrealistic output values, and using only one parameter which could indicate the

308 dilution effectiveness of the diluent.

309 As it will be illustrated later, the dilution effect, i.e. the dilution effectiveness, is found to  
310 vary with the equivalence ratio according to a quadratic polynomial: the dilution effective-  
311 ness is minimum near the stoichiometry and increases as the equivalence ratio shifts towards  
312 the fuel-lean and fuel-rich sides. Moreover, the dilution effectiveness may also depend on  
313 the initial temperature and pressure, and it must be accounted for in the final expression of  
314  $\mu$ . Therefore, the combination of a quadratic  $\varphi$ -dependence with two power terms for the  
315 T- and P-dependences is proposed,

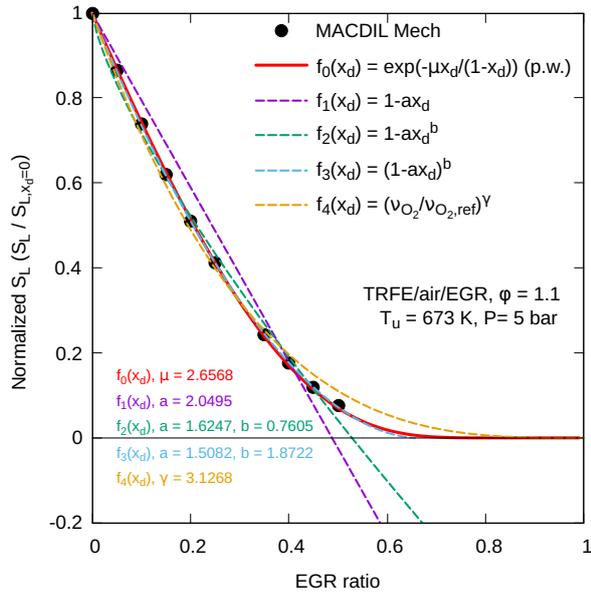
$$\mu = (\mu_0 + \mu_2(\varphi - \varphi_\mu)^2) \left( \frac{T}{T_{ref}} \right)^{\mu_\alpha} \left( \frac{P}{P_{ref}} \right)^{\mu_\beta} \quad (13)$$

316 where  $\varphi_\mu$  is the equivalence ratio where the symmetrical axis of the  $\varphi$ -dependence is located,  
317 and  $\mu_\alpha$  and  $\mu_\beta$  are the power exponents for the T- and P-dependences, respectively.

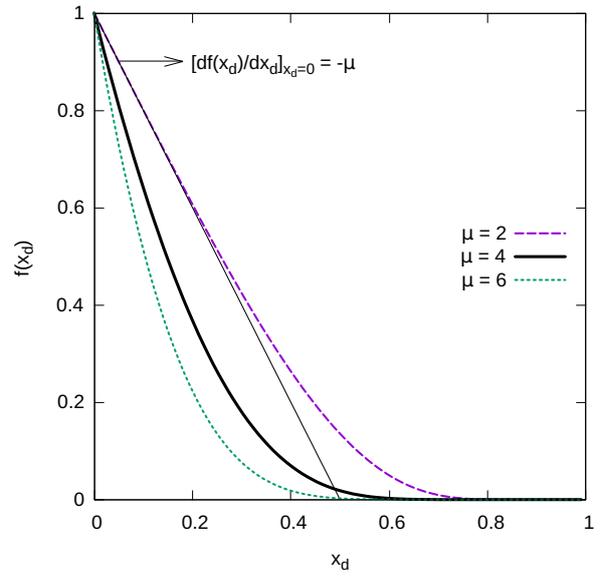
318 Furthermore, for mixture diluents such as EGR, we suggest to estimate the dilution  
319 effectiveness of the mixture  $\mu_{mix}$  based on the dilution effectiveness of each EGR components  
320 using a linear mixing rule,

$$\mu_{mix} = \sum_i X_i \mu_i \quad (14)$$

321 where  $X_i$  and  $\mu_i$  are the mole fractions and dilution effectiveness of the components. The  
322 validity is assessed indirectly based experimental values and discussed later in the paper.



(a) Comparison on fitting performance



(b) Effect of  $\mu$

Figure 4: (a) Comparison between the training computed laminar flame speed data set (MACDIL Mech. computations) and the predictions based on the different dilution term expressions discussed in the main text and (b) effect of changing the dilution effectiveness  $\mu$  on the herein proposed dilution term  $f(x_d)$  (Eq. 12). The reference data used are for the TRFE fuel diluted by synthetic EGR, computed using MACDIL Mech.. (See Section 3)

### 323 **3. Fuel selection and kinetic modeling**

324 Empirical correlations, regardless of their mathematical accuracies, are heavily dependent  
325 on the data used as the training set. Experimental data are scarce and usually limited to  
326 narrow pressure, temperature, and dilution ratio ranges, but validated kinetic models can  
327 help extend these databases. In this section, after introducing the fuel selected for the  
328 present case study, a related kinetic model is assembled and tested against the experimental  
329 data available.

#### 330 *3.1. Choice of Fuel and Available Data*

331 The EGR technology for next-generation engines will be optimized for gasoline fuels, as  
332 many countries in Europe have already enacted the ban of diesel engines and fuel by the  
333 2030s. The present study is part of a larger multi-team effort, MACDIL, aiming at better  
334 characterizing the oxidation and the flame propagation of a particular gasoline fuel under  
335 diluted conditions. This fuel is the commercial gasoline B71 1188 ESSH EURO5 + 20. It  
336 contains 5% ethanol and has a RON and a MON of 96 and 88, respectively. Other properties  
337 of this fuel, reported originally by Di Lorenzo *et al.* [25], are given in Table 2.

338 Di Lorenzo *et al.* [25], as members of the project MACDIL, have experimentally demon-  
339 strated that the commercial gasoline of interest could be emulated by a Toluene Refer-  
340 ence Fuel with Ethanol addition (TRFE) composed of 44% isooctane, 15% n-heptane, 36%  
341 toluene, and 5% ethanol in liquid volume fractions, labeled TAE 7500 gasoline. Table 2  
342 compares the properties of the surrogate and the commercial fuels. Thereafter, we will refer  
343 to this specific surrogate using the abbreviation “TRFE”.

344 Di Lorenzo *et al.* [25] measured the laminar flame speeds of TRFE in air ( $N_2/O_2 =$   
345  $79\%/21\%$  by volume), without and with the dilution of synthetic EGR in a spherical com-  
346 bustion vessel equipped with a double-view Schlieren configuration, enabling to constantly  
347 assess the anisotropy of the propagating flame. The synthetic EGR composition was set to  
348  $74.16\% N_2$ ,  $13.62\% CO_2$ , and  $12.22\% H_2O$  on a molar basis. The authors defined the EGR  
349 ratio as  $X_{EGR}/(X_{fuel} + X_{air} + X_{EGR})$ , and this definition has been adopted hereafter. Data  
350 were collected over a large set of conditions: equivalence ratios ranging from 0.85 to 1.40,

Table 2: Comparison on the properties and composition of one commercial gasoline and the TRFE surrogate. Reproduced from Ref. [25].

Gasoline (B71 1188 ESSH EUROS + 20)		TRFE surrogate (for TAE 7500 gasoline)	
<i>properties:</i>			
RON	96.6	RON	95
C/H/O (mass%)	85.1/13.1/1.8	C/H/O (mass%)	84.79/13.34/1.87
$\rho$ (kg/m <sup>3</sup> ) (at 15 °C)	753.0	$\rho$ (kg/m <sup>3</sup> ) (at 15 °C)	750.5
<i>composition (liq.vol.%):</i>			
ethanol	5.0	ethanol	5.0
n-heptane	3.6	n-heptane	15.0
isooctane	50.0	isooctane	44.0
aromatics (benzene)	33.7 (0.1)	toluene	36.0
olefin	6.2		
methyl-cyclohexane	1.5		

351 temperatures of 373, 423, and 473 K, pressures between 1 and 5 atm, and EGR ratios of 0,  
352 10, and 20%. Though comprehensive, experimental apparatus limitations prevented reach-  
353 ing higher temperatures and pressures that can be encountered in the engine combustion  
354 chamber. Further details on the experimental measurements and the exact content of the  
355 database can be found in [25].

### 356 3.2. Development of the Reduced Model

357 The surrogate fuel selected for this study is a mixture of n-heptane, isooctane, toluene,  
358 and ethanol. Kinetic models for gasoline fuels encompassing these four components have  
359 already been published. Especially, Mehl *et al.* [34] released such a kinetic model, referred  
360 hereafter as LNLL2011, validated on speciation data obtained in flow reactors and igni-  
361 tion delay times. However, this mechanism had not been initially developed for gasoline  
362 surrogates with ethanol and tested against laminar flame speeds. Over the last decade,

no updates have been proposed for this mechanism, especially for the pressure-dependent and third-body reactions that may be sensitive in highly-diluted conditions. In addition, advances have been seen since then on detailed kinetic mechanisms for the aforementioned fuel components as well as core C<sub>0</sub>–C<sub>4</sub> mechanisms for hydrocarbon combustion. Therefore, rather than modifying and updating an existing model, the decision was made to assemble a new kinetic model that would include both the low-temperature and high-temperature chemistries of the four molecules in TRFE.

Aramco Mech 3.0 [35], a comprehensive mechanism for the oxidation of C<sub>0</sub>–C<sub>4</sub> hydrocarbons, has been adopted as the core sub-mechanism. However, the ethanol subset has been substituted by the one published by Zhang *et al.* [36]. The sub-mechanisms of n-heptane (and related species) from Zhang *et al.* [37], of isooctane from Atef *et al.* [38], and of toluene from Yuan *et al.* [39, 40] were then added to the core reaction set. Care was taken during the merging of the different subsets to remove duplicate species and reactions, and to homogenize the species nomenclature. Identified missing reactions were adopted from the LLNL 2011 mechanism. Thermochemical and transport data for each species were inherited from their parent mechanisms. The list of sub-mechanisms used in the initial construction of the detail mechanism and their relative priorities is shown in Table 3.

Table 3: List of sub-mechanisms and their priorities during the construction of the detail mechanism.

Priority <sup>a</sup>	Sub-mechanism	Main Fuel	Ref.
1	LLNL2011	Gasoline Surrogate	[34]
2	Zhang <i>et al.</i> 2016	n-heptane	[37]
3	Atef <i>et al.</i> 2017	Isooctane	[38]
4	Yuan <i>et al.</i> 2015	Toluene	[39, 40]
5	Aramco Mech. 3.0	C <sub>0</sub> –C <sub>4</sub>	[35]
6	Zhang <i>et al.</i> 2018	Ethanol	[36]

<sup>a</sup>The parameters for duplicated species and reactions are replaced by the parameters from the sub-mechanism with the highest value in priority.

380 Despite that Aramco Mech 3.0 is a recent model, rate coefficients of some reactions  
381 potentially sensitive for laminar flame speeds predictions have been the focus of even more  
382 recent publications. Hence, the original rate constants for these reactions were updated.  
383 They mainly consist in reactions of the C<sub>2</sub>-C<sub>3</sub> sub-mechanism. For example, Xiong *et al.*  
384 *al.* [41] and Mai *et al.* [42] investigated theoretically the reaction of ketyl radicals with  
385 hydroxyl radicals (HCCO + OH). They both found that the reaction can proceed either  
386 on a singlet or triplet potential energy surface, and concluded that HCOH (either singlet  
387 or triplet) is the main reaction outcome. Singlet HCOH then further dissociates into H<sub>2</sub> +  
388 CO or undergoes an isomerization into formaldehyde (CH<sub>2</sub>O). Xiong *et al.* derived high-  
389 pressure-limiting rate constants for the different paths identified, and they were included in  
390 the present model. Likewise, the reactions of ketene with free radicals, such as OH, CH<sub>2</sub>, and  
391 CH<sub>3</sub>, are receiving growing interest because of their potential significance to flame speeds  
392 [43, 44] and have thus been the subject of several recent articles [45–47], the conclusions of  
393 which have been included in the detailed model. Further rate constants have been updated  
394 and the reader is invited to refer to the detailed description presented in Section S3 of the  
395 *Supplementary Material*. The model is also provided in the *Supplementary Material* for  
396 further details.

397 At this stage, the kinetic model consists of 2339 species involved in 9440 reactions,  
398 which is prohibitive to perform flame speed calculations. In order to lower the cost of  
399 one-dimensional calculations while maintaining the model accuracy, a reduction procedure  
400 based on adiabatic constant-volume 0-D homogeneous reactor is applied. By selecting a  
401 high initial temperature and target properties relevant to flame speeds (heat release and  
402 concentration of small radicals), this strategy (i) removes low temperature (ii) while retain-  
403 ing high-temperature chemistry important for laminar flame speed calculations (iii) over a  
404 large range of temperature and pressures encountered at later times in the reactor. The  
405 Direct Relation Graph method with Error Propagation (DRGEP) followed by a full sen-  
406 sitivity analysis is applied for the reduction procedure, as implemented in Chemkin Pro  
407 software [48]. TRFE/air/EGR mixtures are considered for the reactants and the matrix of  
408 initial conditions is as follows:  $\varphi = 0.8-1.1-1.4$ ,  $T_i = 1000$  K,  $P_i = 1-5$  bar and EGR ratio =

409 0-20-50%. This matrix, applied to zero-dimensional constant volume calculations, covers a  
410 wide range of temperature (1000–3000 K) and pressure (1-17 bar) to emulate the conditions  
411 encountered throughout the reaction zone of 1-D flames, where the high-temperature com-  
412 bustion chemistry relevant to flame conditions is favored. Target properties were ignition  
413 delay time (IDT), maximum heat release rate, maximum integrated heat release, maximum  
414 mole fractions for H, OH, HO<sub>2</sub>, and CH<sub>3</sub> radicals, with a maximum error tolerance of 5%.  
415 A list of the parameters used during the DRGEP reduction process is presented in the *Sup-*  
416 *plementary Material*, as well as a comparison between the ignition delay times computed  
417 by the detailed and reduced mechanisms. It shows that the reduction strategy succeeded  
418 in retaining only the high temperature chemistry relevant to flames. This final reduced  
419 mechanism encompasses 593 species and 3698 reactions, and is referred to as the MACDIL  
420 Mech. thereafter.

### 421 3.3. Validation of the Reduced Model

422 Before using MACDIL Mech. to extend the database published by Di Lorenzo *et al.*  
423 for the TRFE surrogate fuel, it is tested and validated against experimental laminar flame  
424 speeds measurements available in the literature for TRF and TRFE fuels.

425 The PREMIX code [49] included in the CHEMKIN Pro software [48] was employed to  
426 perform the laminar flame speed calculations. A 20 cm wide domain and fine grid control  
427 parameters (GRAD=0.1 and CURV=0.1) were used to ensure freely propagating flames as  
428 well as grid-independent results. Thermal diffusion (Soret effect) was included. Mixture  
429 average transport was used instead of multi-component transport to reduce computational  
430 time. Comparisons have been performed (included in the *Supplementary Material*) and  
431 showed that the simplified mixture-average calculations are in very close agreement with  
432 the multi-component flame speeds, with absolute differences ranging from 0.01 to 0.57 cm/s  
433 and relative differences ranging from 0.16% to 2.67%.

#### 434 3.3.1. Validation on Pure Component and TRF Mixtures

435 In a first attempt to validate the model, the ability of MACDIL Mech. to predict the  
436 laminar flame speeds of the four components has been assessed. These comparisons are

437 provided in the *Supplementary Material*. The model exhibits an overall good agreement for  
 438 the four fuels, although laminar flame speeds predicted for isooctane and ethanol are close  
 439 to the lower bound of the experimental uncertainties.

440 Figure 5a displays the comparison between the present predictions and the experimen-  
 441 tal data of Manna *et al.* [50] obtained for a TRF surrogate at different initial pressures  
 442 and in a wide equivalence ratio range (0.8–1.6). The experimental data were obtained by  
 443 monitoring the outward propagation of spherical flames in a heated combustion vessel. The  
 444 TRF surrogate used for the experiments has a composition (on a liquid volume basis) of  
 445 77.4% isooctane, 17.6% n-heptane, and 5% toluene. For equivalence ratios between 0.8 and  
 446 1.3, MACDIL Mech. predictions agree well with the experimental value, with deviations  
 447 less than 2 cm/s. At very fuel-rich conditions, predicted flame speeds are significantly lower  
 448 than the experimental observations, up to 5 cm/s at  $\varphi = 1.6$ . Such deviations suggest that  
 449 the MACDIL mechanism can be further improved, although these disparities may also be  
 450 attributed to some extent to experimental uncertainties, especially on the equivalence ratio.

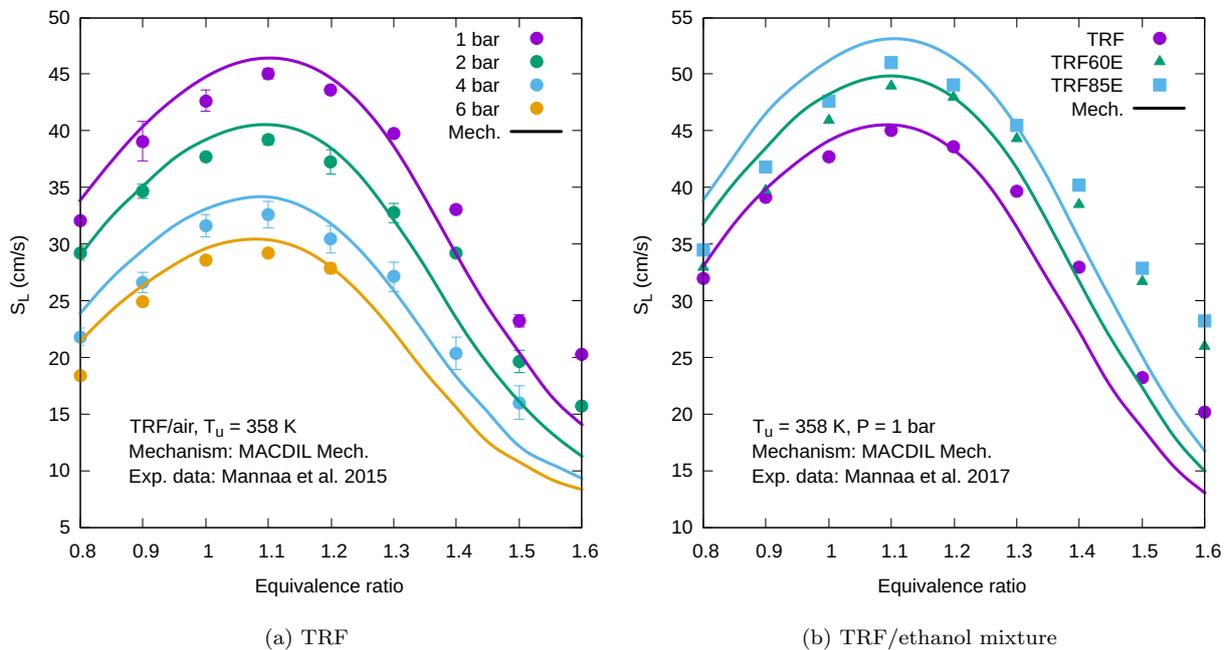


Figure 5: Laminar flame speeds for a TRF surrogate. (a) Effect of pressure [50], and (b) effect of ethanol addition [51]. Experimental data are displayed as symbols and MACDIL Mech. calculations as solid lines.

451 Manna *et al.* [51] also investigated the effect of blending ethanol to the TRF, and they  
452 observed that the laminar flame speed increases with the ethanol amount, regardless of the  
453 equivalence ratio. A similar trend is predicted by MACDIL Mech., as it can be seen in  
454 Figure 5b. Moreover, the agreement between the experimental values and the predicted  
455 ones around the stoichiometry is consistent with the observations without ethanol addition  
456 (deviations less than 5 cm/s) whereas large discrepancies (up to 10 cm/s) exist for very fuel-  
457 rich mixtures with high amounts of ethanol. In their study, Manna *et al.* [51] performed  
458 kinetic modeling using the LLNL2011 mechanism and observed similar deviations at the  
459 fuel-rich conditions.

### 460 3.3.2. Validation on the Targeted TRFE Surrogate

461 The fundamental laminar flame speeds of the target TRFE surrogate have been measured  
462 by Di Lorenzo *et al.* [25] in a spherical vessel, and the influence of four parameters on  $S_L$   
463 has been evaluated: equivalence ratio, temperature, pressure, and EGR ratio.

### 464 Effect of Temperature and Equivalence Ratio

465 The predictions of MACDIL Mech. are compared to the experimental values at atmo-  
466 spheric pressure and three different inlet temperatures (373, 423, 473 K) without EGR in  
467 Figures 6a–6c. For reference, the predictions of the LLNL 2011 [34] model are also dis-  
468 played. For the three temperatures considered, the flame speeds computed with MACDIL  
469 Mech. are close to the measured values, within 5 cm/s (or 10% on a relative basis), for  
470 equivalence ratios up to 1.2. Beyond  $\varphi = 1.2$ , the predicted  $S_L$  are systematically lower  
471 than the experimental ones. The measurements do not exhibit a clear maximum flame  
472 speed since Di Lorenzo *et al.* [25] reported almost similar laminar flame speeds for equiv-  
473 alence ratio between 1 and 1.2 at the three temperatures. On the contrary, the present  
474 mechanism computes the maximum flame speed for equivalence ratio around  $\varphi = 1.1$ , which  
475 is in line with experimental observations for the pure components. Although both models  
476 are in qualitative agreement, the LLNL 2011 computed flame speeds are 2 to 5 cm/s larger  
477 than MACDIL Mech. predictions, and therefore significantly higher than the experimental  
478 values for equivalence ratios below 1.2.

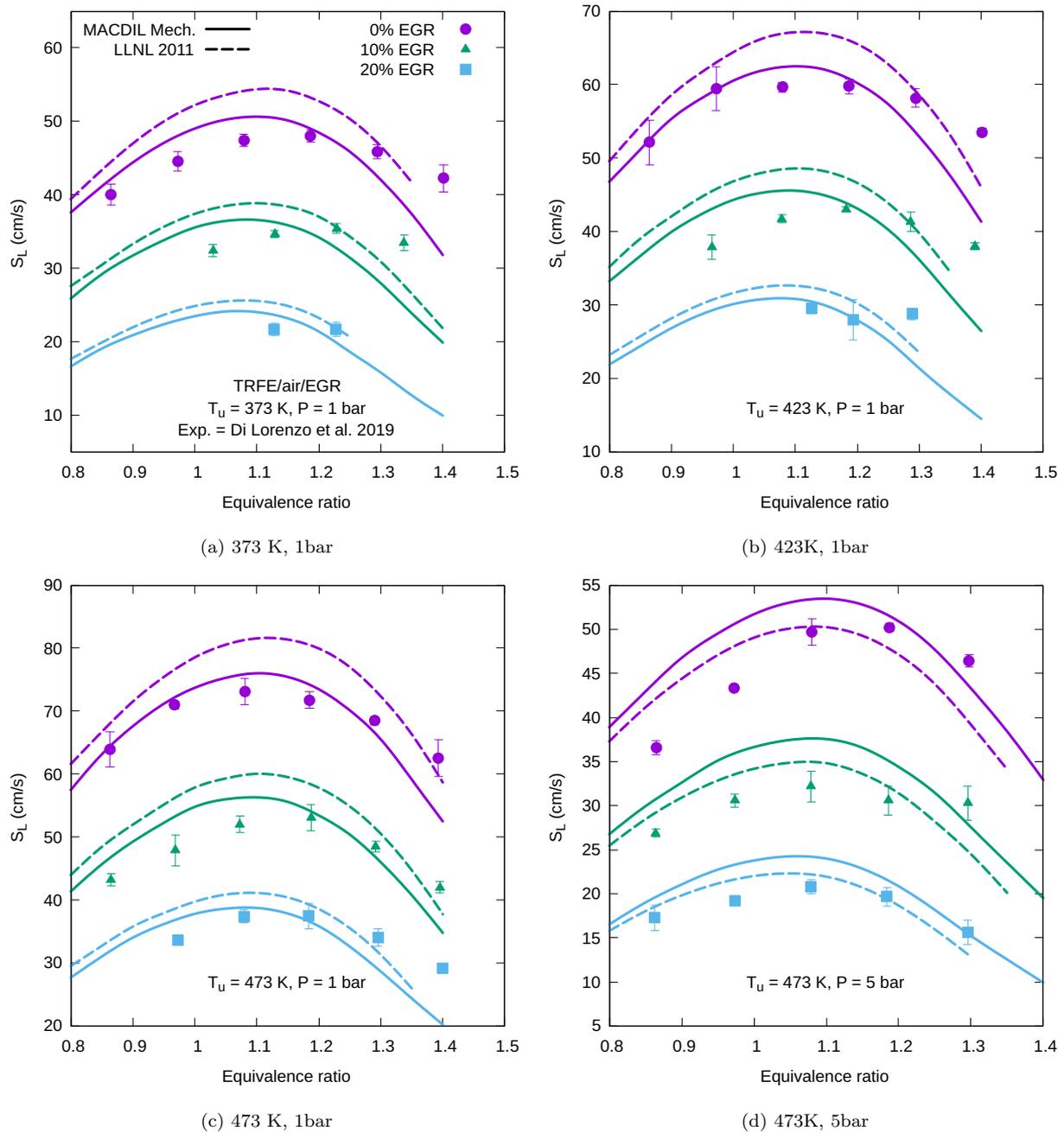


Figure 6: Laminar flame speed of TRFE/air/EGR mixtures: (a) 373 K and 1 bar; (b) 423 K and 1 bar; (c) 473 K and 1 bar; (d) 473 K and 5 bar. Experimental data [25] are displayed as symbols, the MACDIL Mech. and the LLNL model [34] calculations are displayed as solid and dashed lines, respectively.

479 Figure 7 depicts the temperature dependence of the laminar flame speed at an equivalence  
 480 ratio of around 1.1. As expected, the laminar flame speeds increase with the temperature,

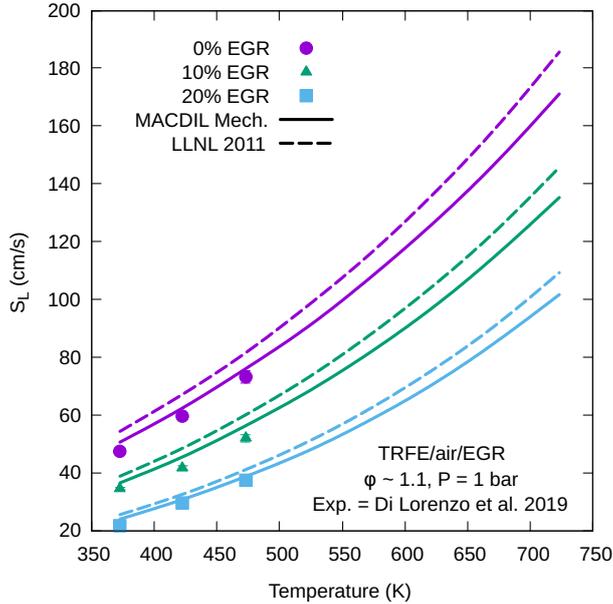


Figure 7: Effect of temperature on the laminar flame speed of TRFE/air and TRFE/air/EGR mixtures at 1 bar and  $\varphi \approx 1.1$ . Experimental data [25] are displayed as symbols, the MACDIL Mech. and the LLNL model [34] calculations are displayed as solid and dashed lines, respectively.

481 which is well captured both qualitatively and quantitatively by the two models. As al-  
 482 ready observed in Figures 6a–6c, the MACDIL Mech. predictions are nevertheless in better  
 483 agreement with the experimental values. While the limited set of data suggests a linear de-  
 484 pendence on the temperature, the modeling data unambiguously show that an exponential  
 485 dependence is most likely. Using the apparent linear dependence would lead to a laminar  
 486 flame speed of 130 cm/s for the TRFE/air mixture, whereas the model prediction is close  
 487 to 160 cm/s, resulting in a deviation of nearly 20%.

#### 488 **Effect of Pressure**

489 Di Lorenzo *et al.* investigated the effect of an increase of the pressure on the fundamental  
 490 laminar flame speed of the TRFE surrogate. The pressure was varied between 1 and 5 bar.  
 491 Figure 6d compares the experimental measurements at 473 K with the model predictions at  
 492 the upper pressure. Unlike the observation at atmospheric pressure, MACDIL Mech. over-  
 493 estimates the laminar flame speeds on the fuel-lean side whereas a much better agreement

494 is obtained for  $\varphi = 1.2$  and  $1.3$ . The maximum flame speed is still predicted by the model  
 495 to occur around  $\varphi = 1.1$ , but experimental data rather suggest an optimum at a slightly  
 496 higher equivalence ratio. The LLNL 2011 model qualitatively agrees with the MACDIL  
 497 Mech. computations, but its predicted laminar flame speeds are slower than the MACDIL  
 498 ones even though the opposite trend was observed at atmospheric pressure. This observa-  
 499 tion suggests that the pressure dependence of the laminar flame speed is more pronounced  
 500 in this model. Figure 8 displays the dependence of  $S_L$  to the pressure according to the  
 501 two models and the experimental measurements for near stoichiometric mixtures ( $\varphi \approx 1.1$ ).  
 502 Both experimental and numerical data exhibit a non-linear decay with the pressure, which  
 503 is in line with the asymptotic theory that predicts  $S_L$  to be proportional to  $P^\beta$ , with  $\beta$   
 504 between -1 and -0.5. The experimental trend is well captured by MACDIL Mech., but the  
 505 LLNL 2011 over-predicts the pressure dependence. Hence, while the LLNL 2011 computed  
 506 laminar flame speeds are higher than that by MACDIL Mech., an inversion is detected for  
 507 pressure around 2.5 atm and they are lower above this threshold.

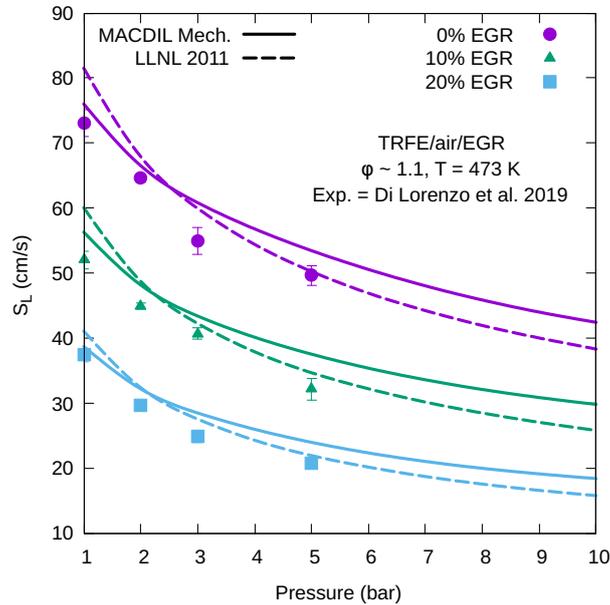


Figure 8: Effect of pressure on the laminar flame speed of TRFE/air/EGR mixtures at 473 K and  $\varphi \approx 1.1$ . Experimental data [25] are displayed as symbols, the MACDIL Mech. and the LLNL model [34] calculations are displayed as solid and dashed lines, respectively.

## 508 Effect of EGR Addition

509 Finally, the ability of MACDIL Mech. to account for the dilution effect on the laminar  
510 flame speed is tested. Di Lorenzo *et al.* [25] measured the laminar flame speeds of TRFE/air  
511 mixtures diluted by 10 and 20% of EGR. The experimental data are compared to the two  
512 mechanisms predictions in Figures 6a–6c. The observations previously made in the undiluted  
513 cases still hold with EGR addition: a maximum flame speed predicted around  $\varphi = 1.1$   
514 whereas experimental values rather suggest  $\varphi = 1.2$ , computed laminar flame speeds are  
515 higher than the experimental data on the fuel-lean side but lower for fuel-rich mixtures, and  
516 the LLNL 2011 predictions are higher than the MACDIL ones. However, the discrepancies  
517 between the two mechanisms shrink with the increase of the EGR dilution ratio, from  $\sim 5$   
518 cm/s at 0% to  $\sim 1.5$  cm/s at 20% for  $\varphi = 1.1$ . This apparent reconciliation between the two  
519 models is actually a scaling effect, since as the dilution ratio increases, the laminar flame  
520 speed, and so the absolute difference between the two models, decreases but the relative  
521 difference remains constant, around 6% at  $\varphi = 1.1$  regardless of the EGR ratio or initial  
522 temperature.

523 Figures 7 and 8 compare the experimental and computed temperature and pressure de-  
524 pendencies of the laminar flame speeds for different EGR dilution ratio at near stoichiometric  
525 conditions. The agreement between the experimental and the computed  $S_L$  improves as the  
526 EGR ratio increases, although MACDIL Mech. slightly overestimates the laminar burning  
527 velocity. Nevertheless, the model computes an overall pressure and temperature depen-  
528 dence that are weakly, if none, sensitive to the EGR dilution ratio, which is consistent with  
529 the experimental observations despite the narrow experimental temperature and pressure  
530 ranges.

531 The effect of the EGR dilution ratio at constant temperature and pressure on the lam-  
532 inar flame speed of TRFE is displayed in Figure 9a. Over the range (0-20%) investigated  
533 by Di Lorenzo *et al.*, both a qualitative and quantitative agreement is observed between the  
534 experimental data and the computations using MACDIL Mech. and LLNL 2011. However,  
535 although the measurements might suggest a linear dependence of the laminar flame with

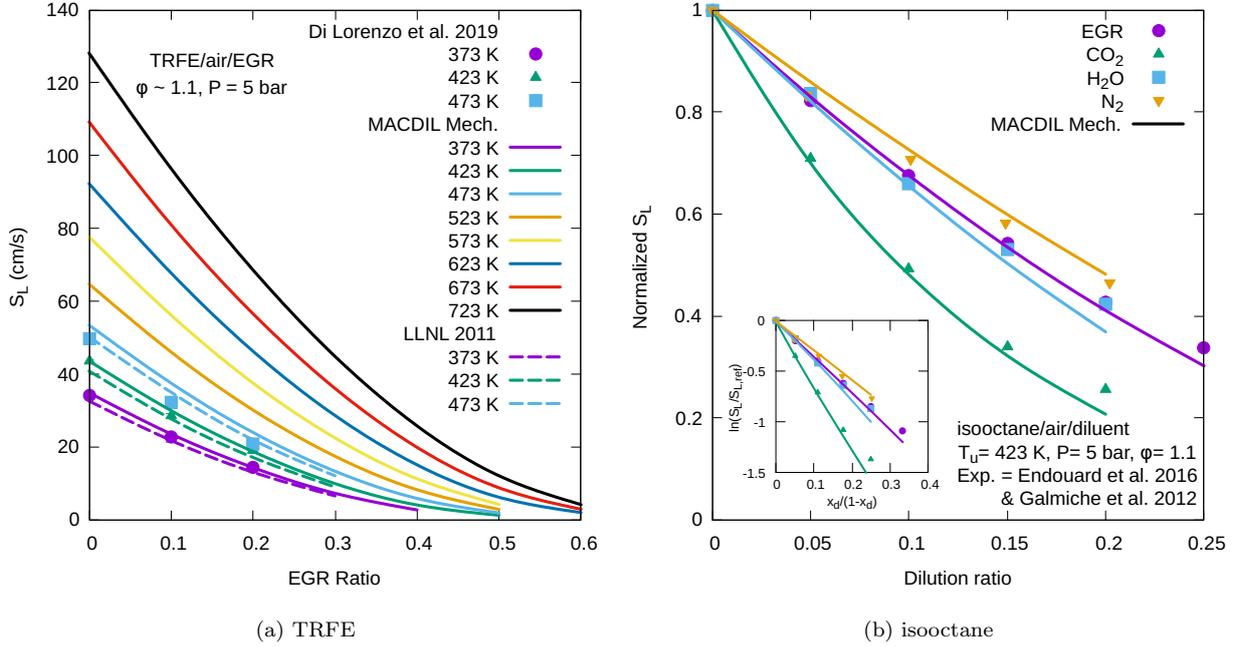


Figure 9: (a) Effect of the dilution ratio  $x_d$  on the laminar flame speeds of TRFE/air/EGR mixtures ( $\phi=1.1$ , 5 bar ; see main text for EGR composition) at different temperatures (symbols are experimental data [25], solid and dashed lines are MACDIL Mech. and LLNL model [34] calculations, respectively). (b) Effect of the diluent composition on the normalized laminar flame speeds of isooctane/air/diluent mixtures ( $\phi=1.1$ , 423 K, 5 bar [24] [23]) ; normalized  $S_L$  is the ratio of the flame speed over the  $S_L$  at a reference condition, i.e.,  $S_L/S_{L,ref}$ , which in this case is the flame speed at the non-diluted condition; the insert of Fig (b) displays the linearity between  $\ln\left(\frac{S_L}{S_{L,ref}}\right)$  and  $\frac{x_d}{1-x_d}$ , with the slope being equal to the reduction effectiveness  $\mu$  of the diluent.

536 the dilution ratio, this observation must be mitigated by the limited set of data and the  
 537 uncertainties associated. The MACDIL mechanism on the contrary predicts a non-linear  
 538 relationship between the two quantities ( $S_L$  and EGR ratio). In spite of the lack of experi-  
 539 mental data at temperatures above 473 K, calculations with MACDIL Mech. are nonetheless  
 540 displayed in Figure 9a as they show that the dilution effect is temperature dependent, with  
 541 a stronger dependence at high temperatures. Indeed, one can clearly see that as the dilu-  
 542 tion ratio increases, the effectiveness of the diluent in reducing the laminar flame speeds  
 543 decreases:  $dS_L/dx_d = -300$ ,  $-220$ , and  $-120$  cm/s for  $x_d = 0.1$ ,  $0.3$ , and  $0.5$ , respectively at  
 544 723 K.

545 The EGR employed by Di Lorenzo *et al.* is a ternary mixture ( $\text{CO}_2/\text{H}_2\text{O}/\text{N}_2$ ), and  
 546 therefore the overall effect observed is a combination of the thermal and chemical effects  
 547 of the three components. Although no data are available for the targeted surrogate fuel  
 548 diluted in each of the EGR components, Galmiche *et al.* [23] and Endouard *et al.* [24]  
 549 carried out such measurements for isooctane/air mixtures, considering dilution in  $\text{N}_2$ ,  $\text{CO}_2$ ,  
 550 and  $\text{H}_2$ . Since the surrogate fuel comprises nearly 50% of isooctane, the measurements of  
 551 Endouard *et al.* can be used to assess the ability of MACDIL Mech. to capture the dilution  
 552 effectiveness of these two diluents. This comparison is displayed in Figure 9b, in which  
 553 the normalized flame speeds (with respect to the non-diluted case) of near stoichiometric  
 554 isooctane/air mixtures are reported as a function of the dilution ratio. It can be seen that  
 555 MACDIL Mech. exhibits good overall agreement with measurements. Consistent with the  
 556 experimental observations,  $\text{CO}_2$  is computed to be the most effective diluent in slowing  
 557 down the flame propagation. While the mitigating effect of  $\text{N}_2$  and  $\text{H}_2\text{O}$  on  $S_L$  is similar  
 558 yet less than the one of  $\text{CO}_2$  according to the experimental measurements, MACDIL Mech.  
 559 computes slightly lower flame speeds for  $\text{H}_2\text{O}$  and slightly higher ones for  $\text{N}_2$ , yielding larger  
 560 difference in dilution effect.

561 From the results in Figure 9b, the dilution effectiveness  $\mu$  of these three molecules can  
 562 be derived by plotting  $\ln(S_L/S_{L,ref})$  as a function of  $x_d/(1 - x_d)$  according to the dilution  
 563 efficiency expression proposed in this study (Eq. 12). The insert in figure 9b shows that a  
 564 linear relationship is indeed obtained, leading to the following effectiveness (in descending  
 565 order) for the three pure diluents:  $\mu_{\text{CO}_2} = 6.06$  (6.45),  $\mu_{\text{H}_2\text{O}} = 3.56$  (3.95), and  $\mu_{\text{N}_2} = 3.10$   
 566 (2.91) (The values in brackets are those derived from the modeling results). With the same  
 567 approach, the dilution effectiveness of the ternary mixture, the synthetic EGR, is close to  
 568 the one of  $\text{H}_2\text{O}$ :  $\mu_{\text{EGR}} = 3.41$  (3.56). These data are summarized in Table 4 along with the  
 569 composition of the EGR mixture.

570 The knowledge of the dilution effectiveness of the three pure diluents and the mixture  
 571 enable assessing the validity of the linear mixing rule proposed earlier (14), at least for  
 572 this mixture and fuel. According to the mixture rule, the dilution effectiveness of the EGR  
 573 mixture is calculated to be 3.56 from the experimental values of the three components, which

Table 4: Dilution effectiveness  $\mu$  (Eq. 14) of several diluents and their mixtures derived from experimental measurements and mechanism predictions (Figure 9b)

Component	Mole Fraction	$\mu_i$	$\mu_{\text{EGR}}$	$\mu_{\text{EGR}}$ estimated by Eq. 14
<i>Experimental Measurements</i>				
N <sub>2</sub>	74.16%	3.10		
H <sub>2</sub> O	12.22%	3.56	3.41	3.56
CO <sub>2</sub>	13.62%	6.06		
<i>MACDIL Mech.</i>				
N <sub>2</sub>	74.16%	2.91		
H <sub>2</sub> O	12.22%	3.95	3.56	3.52
CO <sub>2</sub>	13.62%	6.45		

574 is very close to 3.41 according to the experimental measurements. Using the data based on  
575 the model predictions, an estimation of 3.52 is derived, which is even closer to the value,  
576 3.56, obtained with the mechanism. Therefore, it may be possible to quickly calculate the  
577 dilution effectiveness of an arbitrary diluent mixture, providing the dilution effectiveness of  
578 neat diluents, which could greatly simplify the estimation of  $S_L$  for complex mixtures and  
579 extend the validity range of the empirical correlation. However, further investigations are  
580 still necessary to provide comprehensive proof or validation of this assumption, i.e., linear  
581 mixing rule of dilution effectiveness.

## 582 4. Correlation parameters for TRFE and its Predictive Performance in $S_L$

### 583 4.1. Generation of the database and determination of the correlation parameters

584 The reduced kinetic mechanism MACDIL Mech. described in the previous section has  
585 been used to extend the experimental database of Di Lorenzo *et al.* [25]. The final database  
586 for the laminar flame speeds of TRFE/air/EGR mixtures covers fresh gas temperatures  
587 between 373 and 973 K, pressures from 1 to 50 bar, equivalence ratios from 0.4 to 2.0, and  
588 EGR dilution ratios between 0 and 80%.

589 Correlation parameters were obtained by minimizing the Root Mean Squared Error  
590 (RMSE),  $\sqrt{\frac{\sum(S_{L,i,corr}-S_{L,i,data})^2}{n}}$ . and are given in Table 5. The reference temperature  $T_{ref}$ ,  
591 the reference pressure  $P_{ref}$ , and the reference EGR ratio were arbitrarily fixed at 473 K, 1  
592 bar, and 0%, respectively. Parameters for the reference flame speed (Eq. 9) were determined  
593 first and then held constant in the optimization of the parameters relative to the temper-  
594 ature and pressure exponents and the dilution term. The constraints for each parameter  
595 are also listed in Table 5. Additional constraints were also imposed for the derivative of  
596 flame speeds (positive at over-lean and negative at over-rich conditions) to make sure the  
597 generation of realistic  $S_L$  at these edge conditions. For conditions where both experimental  
598 data and computed flame speeds are available, a weight ratio of 1:1 between the two types  
599 of data was kept. Over the whole  $T$ ,  $P$ ,  $\varphi$ , and  $x_d$  range, the optimized set of correlation  
600 parameters exhibits a RMSE of 4.3 cm/s and a coefficient of determination ( $R^2$ ) of 0.9874  
601 based on 4275 data points.

#### 602 4.2. Performance on Predicting $S_L$

603 Gülder [52] was one of the first to propose an empirical correlation for the laminar flame  
604 speeds of gasoline fuels. The mathematical expression adopted for this correlation is given  
605 in Eq. 15 and the corresponding parameters are given in Table 6. This correlation has been  
606 derived from experimental measurements obtained in a constant volume bomb for isooctane  
607 fuel with the addition of ethanol (0-20%). As such, the correlation does not depend on  $T$ ,  
608  $P$ ,  $\varphi$ , and  $x_d$  but also on the ethanol liquid fraction  $v$ .

$$S_L(\varphi, T, P, x_d) = Z \cdot W \cdot \varphi^\eta \cdot \exp[-\xi_1(\varphi - \xi_2)^2] \left(\frac{T}{T_0}\right)^\alpha \left(\frac{P}{P_0}\right)^\beta f(x_d) \quad (15)$$

609 Later, Yahyaoui [53] adopted the mathematical formulation of Gülder on a ternary mix-  
610 ture representative of commercial gasoline (42.9% isooctane, 13.7% n-heptane, and 43.4%  
611 toluene), and developed a correlation for flame speeds based on kinetic modeling using the  
612 mechanism by Bounaceur *et al.* [54]. In line with the work of Gülder, some parameters of  
613 the correlation depend on the ethanol mole fraction of the fuel. These parameters are given  
614 in Table 6 and were obtained from flame speeds computed with an automatically generated

Table 5: Values of the fitted parameters in Eq. (8)–(13) and their constraints.

Category	Parameter	Unit	Value	Constraint
Reference Term	$T_{ref}$	K	473	positive
	$P_{ref}$	bar	1	positive
	$\varphi_m$		1.0998	positive
	$S_{L,max}$	cm/s	76.6025	positive
	$A_1$	cm/s	67.1132	positive
	$B_1$		0.2877	positive
	$A_2$	cm/s	132.4618	positive
	$B_2$		0.5586	positive
Temperature Term	$\varphi_\alpha$		1.0500	positive
	$\alpha_0$		1.9835	positive
	$\alpha_{A_1}$		1.6730	positive
	$\alpha_{B_1}$		0.4173	positive
	$\alpha_{A_2}$		141.0423	positive
	$\alpha_{B_2}$		4.2414	positive
Pressure term	$\varphi_\beta$		5.1563	positive
	$\beta_0$		-0.0383	<i>negative</i>
	$\beta_2$		-0.0117	<i>negative</i>
Dilution term	$\varphi_\mu$		0.9806	positive
	$\mu_0$		3.0861	positive
	$\mu_2$		4.4256	positive
	$\mu_\alpha$		-0.9319	<i>negative</i>
	$\mu_\beta$		0.0538	positive

615 kinetic model validated on the experimental data of Zhao *et al.* [55] and Jerzembeck *et al.*  
616 [56] for the atmospheric pressure and high pressure (10-25 atm) regimes, respectively.

Table 6: Equations and parameters of the existing correlations used for comparison with the correlation developed in the present work.( $v$  and  $X_{et}$  are the liquid volume fraction and mole fraction of ethanol, respectively ;  $x_d$  is the EGR dilution ratio;  $\nu_{O_2}$  is the oxygen concentration in the non-fuel portion of the mixture.)

Name	Equations and Parameters	Ref.
Güdlér	$S_L(\varphi, T, P, x_d) = Z \cdot W \cdot \varphi^\eta \cdot \exp[-\xi_1(\varphi - \xi_2)^2] \left(\frac{T}{T_0}\right)^\alpha \left(\frac{P}{P_0}\right)^\beta f(x_d) \quad \text{cm/s} \quad [52]$ $T_0=300 \text{ K}, P_0=1 \text{ bar}, W=46.58 \text{ cm/s}, \eta=-0.326, \xi_1=4.48, \xi_2=1.075,$ $Z=1+0.07v^{0.35}, \alpha=1.56+0.23v^{0.46}, \beta=-0.22, f(x_d)=1-2.3x_d$	
Yahyaoui	$S_L(\varphi, T, P, x_d) = Z \cdot W \cdot \varphi^\eta \cdot \exp[-\xi_1(\varphi - \xi_2)^2] \left(\frac{T}{T_0}\right)^\alpha \left(\frac{P}{P_0}\right)^\beta f(x_d) \quad \text{cm/s} \quad [53]$ $T_0=400 \text{ K}, P_0=1 \text{ bar}, W=90.31 \text{ cm/s}, \eta=2.4269, \xi_1=3.154, \xi_2=0.68157,$ $Z=1+0.14892X_{et}^{2.4698}, \alpha=2.236-0.19877X_{et}^{2.4698}, \beta=-0.28327,$ $f(x_d)=1-2.4832x_d-0.0020312x_d^2-0.0041743x_d^3$	
PRISME	$S_L(\varphi, T, P, \nu_{O_2}) = S_{L,ref} \left(\frac{T}{473K}\right)^\alpha \left(\frac{P}{1bar}\right)^\beta \left(\frac{\nu_{O_2}}{\nu_{O_2,ref}}\right)^\gamma \quad \text{cm/s} \quad [25]$ $S_{L,ref}=73.0208+3.5315(\varphi-1.1)-138.1265(\varphi-1.1)^2 \quad \text{cm/s}$ $\alpha=1.7495-0.1010(\varphi-1.1)$ $\beta=-0.2481+0.0232(\varphi-1.1)-1.7739(\varphi-1.1)^2$ $\gamma=3.4143+0.4252(\varphi-1.1)$ $\nu_{O_2}=n_{O_2}/(n_{air}+n_{diluent})$	

617 Another existing correlation is the one proposed by Di Lorenzo *et al.* [25] based on their  
618 experimental measurements. Referred hereafter as the PRISME correlation, its mathemat-

619 ical formulation and associated parameters are given by Eq. 16.

$$\begin{aligned}
 S_L &= S_{L,ref} \left( \frac{T}{473K} \right)^\alpha \left( \frac{P}{1bar} \right)^\beta \left( \frac{\nu_{O_2}}{\nu_{O_2,ref}} \right)^\gamma \quad \text{cm/s} & (16) \\
 S_{L,ref} &= 73.0208 + 3.5315(\varphi - 1.1) - 138.1265(\varphi - 1.1)^2 \quad \text{cm/s} \\
 \alpha &= 1.7495 - 0.1010(\varphi - 1.1) \\
 \beta &= -0.2481 + 0.0232(\varphi - 1.1) - 1.7739(\varphi - 1.1)^2 \\
 \gamma &= 3.4143 + 0.4252(\varphi - 1.1) \\
 \nu_{O_2} &= n_{O_2} / (n_{air} + n_{diluent})
 \end{aligned}$$

#### 620 4.2.1. Equivalence ratio dependence

621 Figure 10 compares the predicted laminar flame speeds using the present and the three  
622 above-mentioned correlations. One can observe that both Gülder and Yahyaoui correlations  
623 overpredict the laminar flame speeds of TRFE at these conditions. At  $\varphi \approx 1.1$ , the value  
624 of  $S_L$  estimated by the Gülder correlation is about 17 cm/s higher than the experimental  
625 measurement, while the value calculated with the Yahyaoui correlation is only about 10  
626 cm/s faster. The strong overestimation by the Gülder correlation can be rationalized by  
627 surprisingly high experimental values used in deriving the correlation parameters. Indeed,  
628 the front flame average propagation speed was extracted from the pressure signal and was  
629 not extrapolated to the zero-stretch condition, yielding large uncertainties in the reported  
630 laminar flame speed. Moreover, the gasoline is emulated by neat isooctane in this correlation,  
631 hence introducing another bias. Scaling down the W parameter of the Gülder correlation  
632 to match the most recent isooctane flame speed measurements at room temperature and  
633 atmospheric pressure gives a value of 27 cm/s at  $\varphi = 1.1$  for the conditions of Fig. 10. The  
634 deviation observed with the Yahyaoui correlation may be related to the different TRFE  
635 and EGR compositions between their work and the present one. The present correlation  
636 and the PRISME correlation are in good agreement with the measurements, which could  
637 be expected as they used the experimental data in their training set. Nonetheless, the  
638 two correlations exhibit different trends with respect to the equivalence ratio. The PRISME  
639 correlation yields a maximum flame speed at a slightly higher equivalence ratio (1.1 vs. 1.07)

640 and on the fuel-rich branch predicts a slower decay in  $S_L$  as the equivalence ratio increases.  
 641 This latter difference is mostly due to the limited set of data at high equivalence ratio used  
 642 by Di Lorenzo *et al.* in deriving their correlation, whereas the present work benefited from  
 643 modeling results, thus an extended equivalence ratio range.

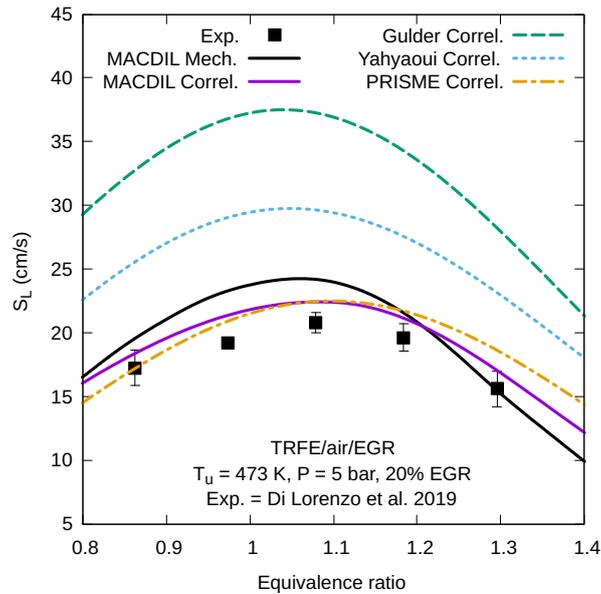


Figure 10: Laminar flame speeds of TRFE/air mixtures at 473 K, 5 bar and diluted in 20% EGR: comparison between experimental data [25], MACDIL Mech. calculations, literature correlations [52, 53, 25] and the present correlation.

#### 644 4.2.2. Effect of Temperature and Pressure

645 Figure 11a displays the temperature dependence of the laminar flame speed according  
 646 to the four correlations as well as the experimental and computed (MACDIL Mech.)  $S_L$   
 647 without EGR addition for near stoichiometric fuel/air mixtures at atmospheric pressure. The  
 648 Gülder and the Yahyaoui correlations overestimate the experimental and computed data on  
 649 the whole temperature range considered (373-973 K), which is consistent with the previous  
 650 observations of Fig. 10. The PRISME correlation is in excellent agreement with the low-  
 651 temperature measurements, but its predictions are lower than the MACDIL computations  
 652 above 600 K. The present correlation, as expected, matches the MACDIL Mech. predictions.  
 653 To unambiguously compare the temperature dependency in the different correlations, the

654 data of Fig. 11a are normalized with respect to the value at the reference temperature  
 655 (473 K) to cancel out other discrepancies related to the reference flame speed expression  
 656 and displayed in Fig. 11b. At low temperatures (373-473 K) where experimental data  
 657 are available, the four correlations exhibit a similar trend, all in good agreement with the  
 658 experimental observations. Deviations start appearing at higher temperatures ( $> 600$  K).  
 659 The Gülder correlation is the one predicting the slowest increase of  $S_L$  with the temperature,  
 660 followed by the PRISME correlation. The MACDIL and the Yahyaoui's correlations predict  
 661 nearly identical trends, with a much faster  $S_L$  increase than the two others, resulting for  
 662 both of them at 973 K in a ratio  $\frac{S_L}{S_{L,ref}}$  of about 7 which is 30 to 40% higher than the  
 663 Gülder and PRISME correlation. This higher ratio is consistent with the MACDIL Mech.  
 664 computations, which have been used for these high temperatures in the training set. This  
 665 substantial deviation between the two groups of correlation highlights the need for high-  
 666 temperature data, either from experiments or computations, in deriving accurate empirical  
 667 correlation since the use of increasing EGR dilution ratios in future combustion engines will  
 668 increase the in-cylinder pre-combustion temperature.

669 Figure 12a illustrates the pressure dependence predicted by the four correlations along  
 670 with the experimental and kinetic modeling trends for a TRFE/air mixture diluted by 20%  
 671 EGR at  $\varphi \approx 1.1$ . The Gülder and the Yahyaoui correlations overestimate the laminar  
 672 flame speed regardless of the pressure, whereas the present and the PRISME correlation  
 673 predicted values are much closer to the experimental and computed values with only minor  
 674 differences. Compared with the PRISME correlation, MACDIL correlation predictions are  
 675 closer to experimental values at low pressure ( $< 5$  bar) and closer to computed values at  
 676 higher pressures ( $> 10$  bar). Figure 12b displays the normalized flame speed (by the value  
 677 of  $S_L$  at 1 bar) without EGR and with 20% EGR addition. The use of normalized values  
 678 levels out differences arising from other parameters than the pressure. Except for Yahyaoui's  
 679 correlation, all correlations match the pressure dependence measured by Di Lorenzo *et al.*  
 680 without EGR at low pressure. At higher pressure, MACDIL correlation predictions are in  
 681 between that of Gülder and PRISME and closer to the computed values.

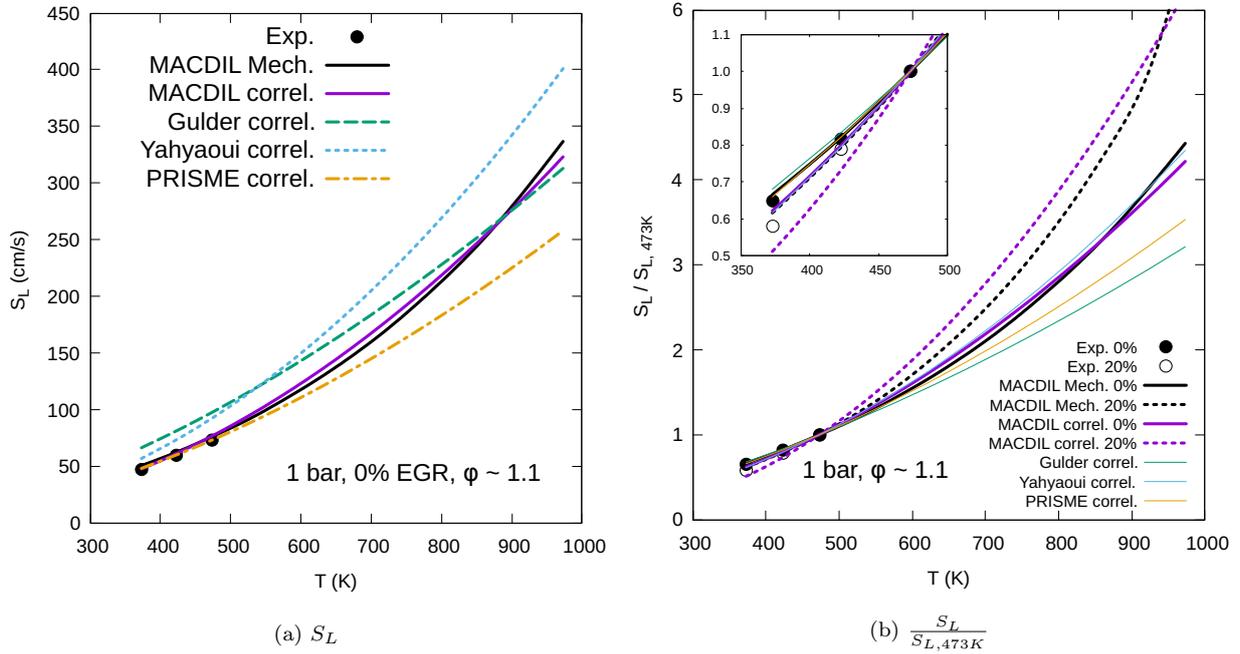


Figure 11: (a) Temperature dependence of the laminar flame speeds of TRFE/air mixtures ( $\varphi \approx 1.1$  and 1 bar) and (b) temperature and dilution ratio dependence of the normalized laminar flame speeds of TRFE/air mixtures: comparison between experimental data [25], MACDIL Mech. calculations, literature correlations [52, 53, 25] and the present correlation.

#### 682 4.2.3. Effect of EGR Addition

683 The effect of the EGR dilution ratio on the predicted laminar flame speeds by the four  
684 correlation is displayed in Fig. 13a at an initial temperature and pressure of 673 K and  
685 5 bar for a near stoichiometric mixture. Such a high temperature was selected to cover  
686 a wide range of dilution ratios, although no experimental data are available. A linear  
687 dependence with the dilution ratio is observed for the Gülder and the Yahyaoui correlations,  
688 resulting in negative laminar flame speeds for dilution ratios above  $\sim 40\%$ . Whereas this  
689 was expected for the Gülder correlation because of the linear function adopted for  $f(x_d)$ ,  
690 such a trend for the Yahyaoui's correlation is more surprising as a cubic polynomial had  
691 been used. Nonetheless, the coefficients of the square and cubic terms are relatively small,  
692 which may result from the lack of experimental data at high dilution ratios in the training  
693 set employed by the authors. The PRISME correlation predicts a dilution ratio dependency

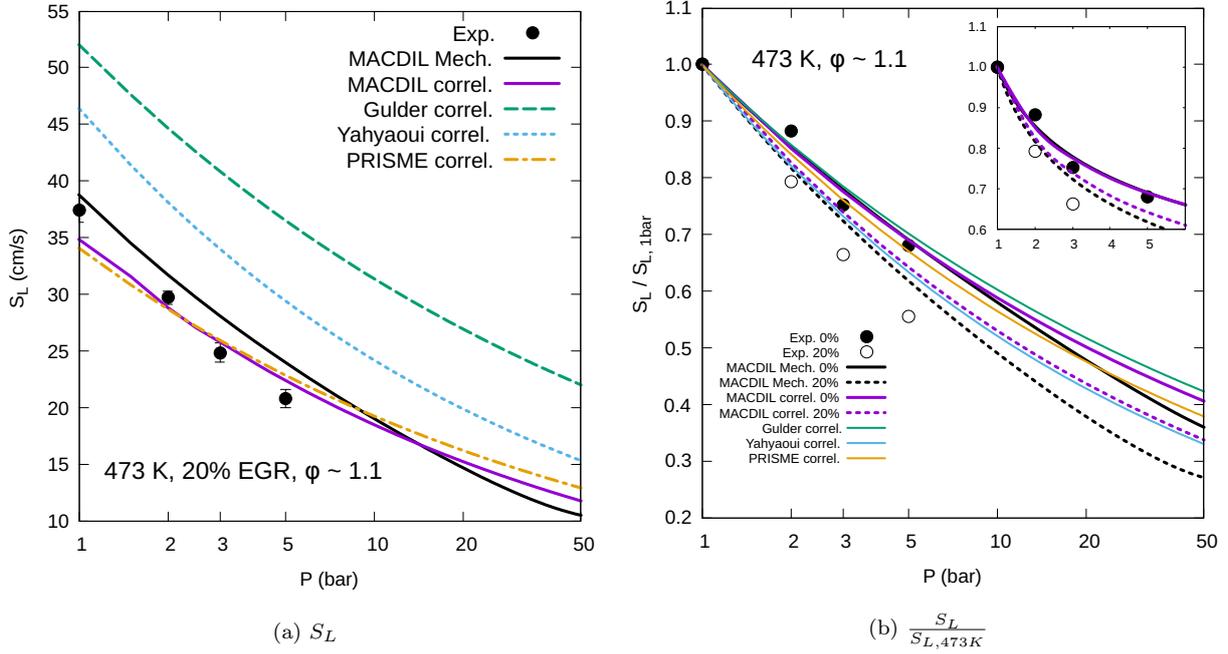


Figure 12: (a) Pressure dependence of the laminar flame speeds of TRFE/air mixtures ( $\phi \approx 1.1$  and 473 K) and (b) pressure and dilution ratio dependence of the normalized laminar flame speeds of TRFE/air mixtures: comparison between experimental data [25], MACDIL Mech. calculations, literature correlations [52, 53, 25] and the present correlation.

694 similar to the mechanism predictions at the lower dilution ratios ( $< 0.2$ ) but underestimates  
 695 the absolute value of flame speeds. At higher dilution ratios ( $> 0.3$ ), the PRISME correlation  
 696 underpredicts the inhibiting effect on the laminar flame speed of the diluent and a weaker  
 697 dilution effect at higher dilutions. The MACDIL correlation exhibits an excellent agreement  
 698 with the computed values over the whole range of dilution ratio, illustrating the advantage  
 699 of the dilution term herein proposed (Eq. 12) over the other expressions.

700 Unlike the Gülder and the Yahyaoui correlations, both the PRISME and the MACDIL  
 701 correlations consider an equivalence ratio dependence of the dilution term, either through  
 702 the exponent (Eq. (16)) or the dilution effectiveness parameter  $\mu$  (Eq. (13)). Figure 13b  
 703 displays the dilution effectiveness  $\mu$  at the reference condition (473 K, 1 bar) derived from  
 704 the experimental data and the MACDIL Mech. calculations as a function of the equivalence  
 705 ratio. A concave quadratic dependency is unambiguously observed in both cases, rational-

706 izing the second-order polynomial expression adopted in Eq. 12. The equivalence ratio for  
 707 which the dilution effectiveness is minimum is obtained at fuel-rich conditions ( $\varphi = 1.20$ )  
 708 for experimental data whereas it is around  $\varphi = 1.05$  according to the MACDIL Mech.. Nev-  
 709 ertheless, the value of this minimum is around 2.6-2.7 in both cases. This figure also shows  
 710 the equivalent dilution effectiveness  $\mu$  obtained when Eq. 12 is applied to the PRISME and  
 711 MACDIL correlations. The PRISME correlation returns a weak linear  $\varphi$ -dependence of the  
 712 dilution effectiveness, which is consistent with the linear expression adopted for the expo-  
 713 nent term of the dilution factor ( $\gamma$  in Eq. 16) but contrary to the experimental observations.  
 714 Differently, the MACDIL correlation exhibits a quadratic  $\varphi$ -dependence. But the minimum  
 715 is observed at a lower equivalence ratio ( $\varphi = 1.00$ ) and the predicted dilution effective-  
 716 ness is consistently higher than computed by the mechanism and experimental values for  
 717 fuel-rich mixtures. It is noteworthy that the MACDIL correlation (Eq. 13) better captures  
 718 the equivalence ratio dependence of the dilution term. The discrepancy observed in Figure  
 719 13b arises to some extent from the uncertainty in the equivalence ratio associated with the  
 720 experimental data (as discussed in the previous section), and the resulting attempt of the  
 721 fitting procedure to conciliate both the experimental and modeling sets of data. A better  
 722 agreement could be achieved by considering only the experimental data or the mechanism  
 723 predictions.

724 In addition to equivalence ratio dependency, the MACDIL Mech. and the experimental  
 725 data indicate that dilution effectiveness could also depend on temperature and pressure and  
 726 vice versa. Figures 11b and 12b provide  $S_L$  evolution with temperature and pressure at  
 727 both 0% and 20% EGR. Despite the limited temperature and pressure ranges investigated,  
 728 experimental measurements indicate that the T and P dependence depends on the dilution  
 729 ratio, which is confirmed by the MACDIL Mech. computations. Therefore, this effect  
 730 must be included in the final correlation. The dilution ratio dependence of the temperature  
 731 and pressure dependences can also be formulated as a T and P dependence of the dilution  
 732 effectiveness, thus preventing the modification of two terms but only one. This was achieved  
 733 by introducing a simple T and P power dependence in the dilution effectiveness expression,  
 734 Eq. 12. The fitting process gives for  $\mu_\alpha$  and  $\mu_\beta$  the values -0.9319 and 0.0538, respectively,

735 implying that increasing the temperature lowers the dilution effectiveness and therefore  
 736 the reduction of the laminar flame speed whereas an increase in pressure enhances the  
 737 EGR reduction effect. It can be seen in Fig. 11b that the fitted value of  $\mu_\alpha$  enables the  
 738 MACDIL correlation to adequately match the MACDIL Mech. computations over the whole  
 739 temperature range. However, as displayed in Fig. 12b, the optimized value of  $\mu_\beta$  results in  
 740 an underestimation of the pressure effect on the dilution effectiveness, and then on the flame  
 741 speed reduction due to the combined effects of the dilution and pressure. Further work is  
 742 thus necessary to improve the correlation performance and its pressure dependence.

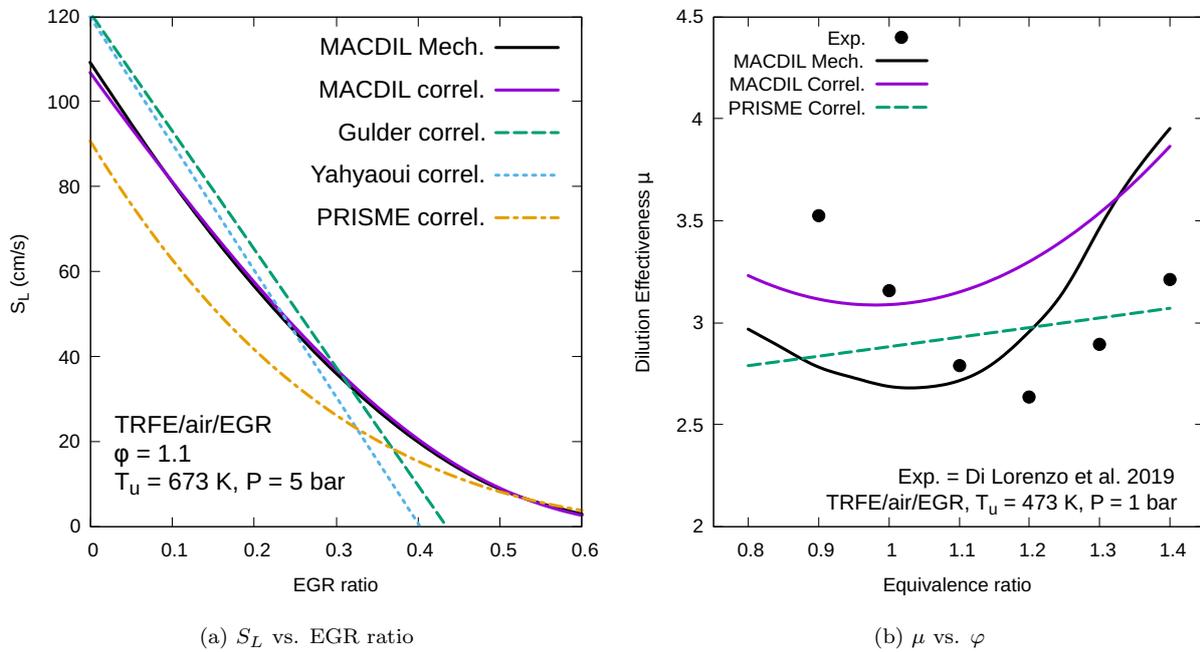


Figure 13: (a) EGR dilution ratio dependence of the laminar flame speeds of TRFE/air mixtures ( $\varphi \approx 1.1$ , 673 K and 5 bar): comparison between experimental data [25], MACDIL Mech. calculations, literature correlations [52, 53, 25] and the present correlation. (b) Equivalence ratio dependence of the dilution effectiveness  $\mu$ .

## 743 5. Conclusion

744 The laminar flame speeds of a TRFE surrogate and their dependences on parameters such  
745 as equivalence ratio, temperature, pressure, and dilution ratio have been investigated. The  
746 limited set of experimental data for TRFE/air EGR mixtures motivated the development of  
747 a reduced mechanism, referred to as the MACDIL Mech., that contains 593 species involved  
748 in 3698 reactions. It was formulated by compiling state-of-art sub-mechanisms from the  
749 literature for each surrogate component (isooctane, n-heptane, toluene, and ethanol), with  
750 updates for key reaction rate constants based on recent theoretical and experimental studies.  
751 The resulting model was finally reduced using a DRGEP-SA method based on 0-D compu-  
752 tations and then successfully validated against the experimental measurements available for  
753 the TRFE/air EGR mixtures in the whole  $[\varphi, T, P, x_d]$  domain. The MACDIL Mech. was  
754 then used to expand the database of laminar flame speeds for the TRFE surrogate of inter-  
755 est to higher temperatures, pressures, and dilution ratios. This newly expanded database  
756 (including both experimental and computed data) was next employed as a training set to  
757 derive the optimized coefficients of a new empirical correlation, the MACDIL correlation.

758 The formulation of this new correlation has been herein devised, with the following new  
759 features:

- 760 1. the laminar flame speed reference term adopts a dual Gaussian formulation in which  
761 the different parameters are related to physical quantities along the  $S_L$ - $\varphi$  curve, thus  
762 enabling to capture the different  $\varphi$ -dependences of the fuel-lean and fuel-rich branches;
- 763 2. the temperature exponent formulation similarly adopts a two-Gaussian expression,  
764 preventing unrealistic overestimations of the laminar flame speeds at over-lean and  
765 over-rich conditions, i.e. near the flammability limits;
- 766 3. the dilution factor is modeled by an exponential expression, enabling a smooth and  
767 non-linear decay of the laminar flame speeds with the dilution ratio and avoiding  
768 negative laminar flame speeds in highly diluted conditions by introducing only one  
769 new parameter, namely the reduction effectiveness  $\mu$ ;

770 4. the dilution effectiveness not only depends on the equivalence ratio ( $\varphi$ ), the tempera-  
771 ture, and pressure (T and P) according to a quadratic expression or a simple exponent  
772 dependence but also on the nature of the diluent (the dilution effectiveness of a mixture  
773 can be calculated based on a linear mixing rule i.e.,  $\mu_{mix} = \sum X_i \mu_i$ ).

774 The proposed correlation, optimized for the TRFE fuel considered, shows an overall  
775 good performance in predicting the dependence of laminar flame speeds to the dilution ratio,  
776 equivalence ratio, temperature, and pressure and outperforms previous literature correlations  
777 for this fuel or similar fuels. Some improvements of the correlation formulation are still  
778 required, especially for the dilution ratio/pressure combined effect on the laminar flame  
779 speed. Likewise, the optimization of the correlation parameters would benefit from a larger  
780 experimental database, covering a larger T, P, and dilution ratio range.

781 Nevertheless, the new correlation can be implemented in CFD codes to provide more  
782 accurate estimations of laminar flame speeds than existing correlations, and thus improve  
783 simulations of turbulent flames at highly EGR-diluted regimes.

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### Semi-Detailed Mechanism

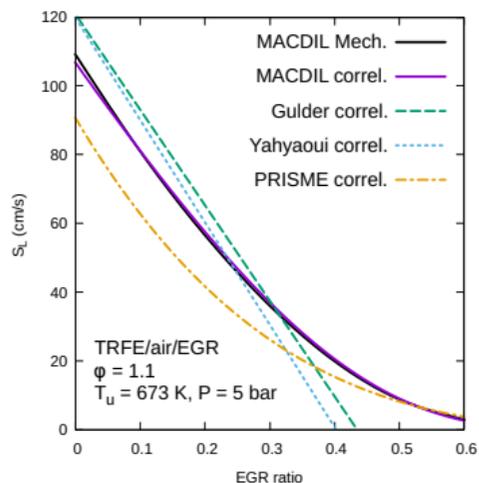
- MACDIL Mech.
- 593 species, 3698 reactions
- Validated on Exp. data

$S_L$  Database  
for TRFE  
Gasoline  
Surrogate

Exp. from Di Lorenzo et al.

### SL Correlation for TRFE

- 4275 dpts: 373-973 K, 1-50 bar,  $\varphi = 0.4-2.0$ , 0 - 80% EGR
- RMSE = 4.3 cm/s,  $R^2 = 0.9874$



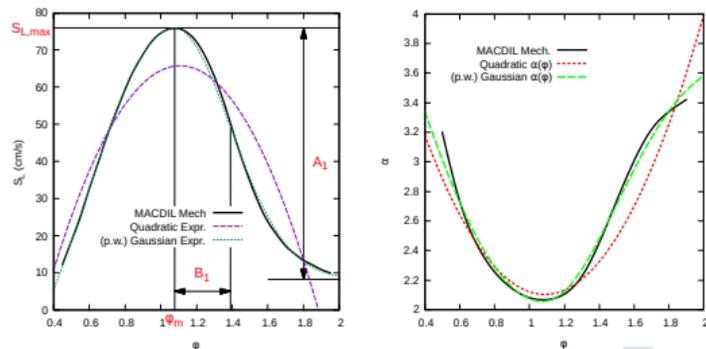
### Novel Formulas for Laminar Flame Speeds Correlation

$$S_L(\varphi, T, P, x_d) = S_{L,ref} \left( \frac{T}{T_{ref}} \right)^\alpha \left( \frac{P}{P_{ref}} \right)^\beta f(x_d)$$

### Reference Flame Speed & Temperature Exponent

- Gaussian function + step function
- Better fit at over-rich and over-lean conditions
- Avoid unrealistic SL near flammability limits

$$S_{L,ref}(\varphi) = \begin{cases} A_1 \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_m}{B_1} \right)^2 \right] + (S_{L,max} - A_1), & \varphi > \varphi_m \\ A_2 \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_m}{B_2} \right)^2 \right] + (S_{L,max} - A_2), & \varphi \leq \varphi_m \end{cases}$$



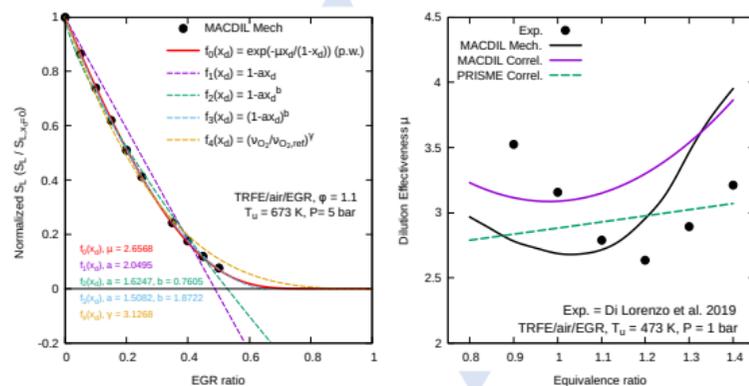
$$\alpha(\varphi) = \begin{cases} \alpha_0 + \alpha_{A_1} - \alpha_{A_1} \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_\alpha}{\alpha_{B_1}} \right)^2 \right], & \varphi > \varphi_\alpha \\ \alpha_0 + \alpha_{A_2} - \alpha_{A_2} \exp \left[ -\frac{1}{2} \left( \frac{\varphi - \varphi_\alpha}{\alpha_{B_2}} \right)^2 \right], & \varphi \leq \varphi_\alpha \end{cases}$$

$$\beta(\varphi) = \beta_0 + \beta_2(\varphi - \varphi_\beta)^2$$

### Dilution Factor

- Exponential formula for non-linear dilution effect
- Better fit at highly-diluted conditions
- Only 1 parameter needed:  $\mu$

$$f(x_d) = \exp \left( -\frac{\mu x_d}{1 - x_d} \right)$$



$$\mu(\varphi, T, P) = (\mu_0 + \mu_2(\varphi - \varphi_\mu)^2) \left( \frac{T}{T_{ref}} \right)^{\mu_\alpha} \left( \frac{P}{P_{ref}} \right)^{\mu_\beta}$$

### Dilution Effectiveness ( $\mu$ )

- Diluent-specific
- Dependent on Equivalence ratio
- Follow linear blending rule

$$\mu_{mix} = \sum_i X_i \mu_i$$